

# Coupled Reconstruction of 2D Blood Flow and Vessel Geometry from Noisy Images via Physics-Informed Neural Networks and Quasi-Conformal Mapping\*

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**Abstract.** Blood flow imaging provides important information for hemodynamic behavior within the vascular system and plays an essential role in medical diagnosis and treatment planning. However, obtaining high-quality flow images remains a significant challenge. In this work, we address the problem of denoising flow images that may suffer from artifacts due to short acquisition times or device-induced errors. We formulate this task as an optimization problem, where the objective is to minimize the discrepancy between the modeled velocity field, constrained to satisfy the Navier-Stokes equations, and the observed noisy velocity data. To solve this problem, we decompose it into two subproblems: a fluid subproblem and a geometry subproblem. The fluid subproblem leverages a Physics-Informed Neural Network to reconstruct the velocity field from noisy observations, assuming a fixed domain. The geometry subproblem aims to infer the underlying flow region by optimizing a quasi-conformal mapping that deforms a reference domain. These two subproblems are solved in an alternating Gauss-Seidel fashion, iteratively refining both the velocity field and the domain. Upon convergence, the framework yields a high-quality reconstruction of the flow image. We validate the proposed method through experiments on synthetic flow data in a converging channel geometry under varying levels of Gaussian noise, and on real-like flow data in an aortic geometry with signal-dependent noise. The results demonstrate the effectiveness and robustness of the approach. Additionally, ablation studies are conducted to assess the influence of key hyperparameters.

**Key word.** Flow Image Reconstruction, Blood Flow Simulation, Physics-Informed Neural Network, Quasi-Conformal Geometry, Image Denoise, Image Segmentation

**AMS subject classifications.** 65N21, 65D18, 76D05, 92C50

**1. Introduction.** Blood flow imaging plays a fundamental role in characterizing cardiovascular dynamics, offering critical insights into hemodynamic behavior within the vascular system. This technique is indispensable for medical monitoring, cardiovascular disease diagnosis, and surgical planning. As a non-invasive approach, it enables safe assessment of patients with severe cardiovascular conditions while avoiding the flow disturbances caused by intravascular instrumentation [41].

However, achieving accurate and high-resolution blood flow imaging remains a significant challenge. Existing modalities, such as flow MRI, often require long acquisition times [26] to produce clear and high-quality flow images. This not only causes discomfort for patients but

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34 also makes the imaging process highly susceptible to motion artifacts, such as those caused  
35 by respiration. On the other hand, fast acquisition approaches such as low-dose flow MRI or  
36 Doppler ultrasound offer faster imaging but typically suffer from high noise levels and signal  
37 loss, resulting in data that may be unreliable for precise diagnostic purposes.

38 In this work, we focus on the problem of reconstructing flow images that are often cor-  
39 rupted by artifacts resulting from short acquisition times or device-induced noise. This task  
40 has been modeled by [21, 22] as an optimization problem, by minimizing the discrepancy  
41 between a modeled velocity field, constrained by the incompressible Navier-Stokes equations,  
42 and the noisy observations.

43 To more effectively tackle this problem, we propose a reformulation that decomposes into  
44 two coupled subproblems: a fluid subproblem and a geometry subproblem. In the fluid sub-  
45 problem, the flow dynamics are reconstructed within a fixed domain by minimizing the discrep-  
46 ancy between the modeled and measured velocity fields, while ensuring physical consistency  
47 through the incompressible Navier-Stokes constraints. In the geometry subproblem, a correc-  
48 tion mapping, interpretable as a domain gradient, is estimated to register the reconstructed  
49 velocity to the observed data, thereby capturing domain misalignments and approximating the  
50 gradient with respect to the spatial domain. By alternating between the two subproblems in  
51 a Gauss-Seidel fashion, the overall optimization proceeds iteratively until convergence, which  
52 then solves the original problem with theoretical support.

53 To efficiently solve this coupled optimization problem while leveraging the advantages of  
54 differentiable architectures, we design a new algorithm by making the entire framework to be  
55 network-based. For the fluid subproblem, a novel Physics-Informed Neural Network (PINN)  
56 is used to reconstruct the velocity field, allowing the Navier-Stokes constraints to be enforced  
57 directly through backpropagation. For the geometry subproblem, a U-Net architecture is  
58 adopted to predict the quasi-conformal mapping, which iteratively corrects geometric mis-  
59 alignments by minimizing the discrepancy between the measured and reconstructed velocity  
60 fields through a correction mapping.

61 We validate our method through comprehensive experiments. First, we evaluate perfor-  
62 mance on synthetic flow images in a converging channel geometry under both low and high  
63 Gaussian noise levels. Next, we test the approach on more anatomically realistic aorta geome-  
64 tries using a signal-dependent noise model. In all cases, our method produces more accurate  
65 and higher-quality reconstructions compared to conventional baselines. We also performed  
66 ablation studies to assess the impact of different weighting parameters in the optimization  
67 of the fluid and geometry subproblems, identifying effective hyperparameter choices. Note  
68 that our work is presented and implemented in the 2D setting; however, it can be potentially  
69 extended to 3D by replacing the Beltrami coefficient with 3D conformality measures in [43].

70 Overall, the contribution of this task could be summarized as follows.

- 71 • We propose a novel framework for jointly reconstructing both the blood flow field and  
72 the flow domain geometry from noisy velocity images, by integrating the Navier-Stokes  
73 model with quasi-conformal mapping theory.
- 74 • We reformulate the inverse Navier-Stokes problem into two decoupled subproblems:  
75 a fluid subproblem and a geometry subproblem. Theoretical analysis is provided to  
76 justify the validity of this decomposition.
- 77 • We develop a method that utilizes Physics-Informed Neural Networks to reconstruct

78 the velocity field by fitting to noisy measurement data and enforcing the Navier-Stokes  
79 equations as soft physical constraints.

- 80 • We introduce a mapping approach for segmenting the flow region using quasi-conformal  
81 geometry, which enables globally guided domain evolution and leveraging morpho-  
82 logical priors from the template.

## 83 2. Related Work.

84 **2.1. Blood Simulation.** Accurate blood flow measurement is vital for cardiovascular  
85 healthcare, informing diagnosis, surgical planning, and treatment. Although invasive meth-  
86 ods such as Fractional Flow Reserve (FFR) [31] are widely used, they pose risks such as  
87 complications and can disrupt the dynamics of natural flow [41], leading to measurement in-  
88 accuracies. Numerical simulations offer a noninvasive alternative by modeling blood as an  
89 incompressible Newtonian fluid governed by the Navier-Stokes equations [30]. To account for  
90 vessel deformation, these models are extended to fluid-structure interaction (FSI) frameworks  
91 [32].

92 Methods have been proposed to solve FSI problems. These include dimensional coupling  
93 [13], parallel preconditioning [11], and splitting schemes with stability guarantees [7]. Other  
94 approaches involve space-time formulations [6], discontinuous Galerkin methods [39], coupled  
95 momentum methods [12], and learning-based approach [44, 48].

96 **2.2. Image Denoise.** Image restoration is challenged by noise types such as speckle, Pois-  
97 son, and Rician noise, which degrade quality and complicate analysis. To address these, a  
98 variety of denoising methods have been developed. Lou et al. [25] proposed a weighted dif-  
99 ference between anisotropic and isotropic total variation (TV) regularization. Li et al. [24]  
100 introduced a variational model for multiplicative noise removal based on the difference of con-  
101 vex functions [35] combined with primal-dual algorithms. Xiao et al. [40] further improved  
102 impulsive noise removal by incorporating adaptive proximal parameters into a nonconvex TV-  
103 log model. To reduce dependence on initialization, Getreuer et al. [16] developed the convex  
104 Getreuer-Tong-Vese (GTV) model for MAP-Rician noise, efficiently solved using Bregman  
105 splitting [17]. Chen et al. [9] proposed the strictly convex CZ model, which adds a quadratic  
106 regularization term for stability.

107 Beyond these convex approaches, more advanced techniques employing higher-order and  
108 nonconvex regularization have also been explored. Kang et al. [18] proposed a spatially adap-  
109 tive nonconvex prior to better preserve fine structures, and Martin et al. [27] demonstrated  
110 effective Rician noise removal using nonconvex fidelity terms in conjunction with proximal  
111 point algorithms. These developments reflect the ongoing effort to improve robustness, struc-  
112 tural preservation, and reconstruction accuracy in challenging noise environments.

113 **2.3. Physics-Based Reconstruction.** The reconstruction of flow images [10], often cor-  
114 rupted by artifacts from short acquisition times or device noise, has been addressed using  
115 physics-based approaches. Kontogiannis *et al.* [21] formulated the inverse Navier-Stokes  
116 problem, which optimizes over the fluid dynamics and the computation domain by minimiz-  
117 ing the discrepancy between modeled velocity fields and observations under incompressible  
118 Navier-Stokes constraints. This was extended to 3D flows in [22], introducing a stabilized  
119 Nitsche cut-cell FEM for high-Reynolds-number flows, an implicit geometry representation

120 with a viscous Eikonal constraint.

121 In parallel, Aguayo *et al.* [3, 4, 2] developed methods for modeling obstacles in incompressible flows using fictitious domains, penalization potentials, and Brinkman-type permeability terms for distributed-resistance treatment. For problems with known geometries, physics-informed neural networks have been applied to reconstruct flow fields: Sun *et al.* [34] proposed a physics-constrained neural network to reconstruct flow fields from sparse and noisy measurements, combining probabilistic modeling of physical constraints with data uncertainty to guide the network toward physically consistent solutions. Gao *et al.* [15] developed a CNN-based framework for super-resolution and denoising of fluid flows, using physics-informed losses to reconstruct high-resolution fields from coarse or noisy observations without high-resolution labels. Wang *et al.* [37] employed physics-informed neural networks regularized by the Navier-Stokes equations for dense flow reconstruction, but their method does not account for boundary conditions or domain estimation during the reconstruction process.

133 The problem investigated in this study is inspired by the work of [21], but our approach introduces several key differences. First, we reformulate the original problem into two alternating solved coupled subproblems explicitly. Such a reformulation differs from previous approaches and is supported by a dedicated theoretical result (Theorem 3.3), ensuring that it is mathematically well founded. Second, we solve the fluid subproblem in a new way by using a physics-informed neural network (PINN), which removes the need for explicit numerical discretization, making the approach meshless and allowing the integration of modern deep learning techniques. Third, the geometry subproblem is newly formulated as a registration task, where the domain evolution is represented by a learned deformation map. Fourth, we incorporate quasi-conformal (QC) theory into the geometric component to ensure topology preservation and naturally embed morphological priors in the reconstruction process.

Notation	Description	Notation	Description
$\boldsymbol{x}$	Spatial point	$c$	Correction mapping
$\boldsymbol{u}$	Modeled flow velocity	$f$	Deformation mapping
$p$	Kinetic pressure	$\Omega_0$	Reference flow domain
$\nu$	Kinematic viscosity	$\Omega$	Flow domain
$\boldsymbol{n}$	Outward normal vector	$\Gamma^i$	Inlet boundary
$\boldsymbol{g}$	Inlet boundary condition	$\Gamma^o$	Outlet boundary
$\tilde{\boldsymbol{u}}$	Measured flow velocity	$\Gamma^w$	Wall boundary

**Table 1**

*Notation table for problem formulation.*

144 **3. Inverse Navier-Stokes problem.** The main objective of this work is to reconstruct the velocity field over the vessel lumen region, given only a measured blood flow image. To do so, the model should not only try to extract information from the measured data but also try to satisfy its fluid physics, which is generally described by the incompressible Navier-Stokes equation [32]. Before presenting the model, we first describe the notation that is needed for the discussion. Let  $D$  denote the full image domain, and let  $\tilde{\boldsymbol{u}}$  represent the noisy image obtained from measurements. We describe the lumen domain of the blood vessel as  $\Omega \subset D$ .

151 Within this domain  $\Omega$ . The boundary of the vessel domain  $\partial\Omega$  is divided into three parts: the  
 152 inlet  $\Gamma^i$ , the outlet  $\Gamma^o$ , and the wall of the vessel  $\Gamma^w$ , satisfying  $\partial\Omega = \Gamma^i \cup \Gamma^o \cup \Gamma^w$ .

153 In this inverse problem setting, only the noisy velocity field  $\tilde{\mathbf{u}}$  is available from the imaging  
 154 data, while the underlying flow domain  $\Omega$ , the inlet boundary condition  $\mathbf{g} \in C(\Gamma^i)$ , and the  
 155 true velocity field  $\mathbf{u}$  are unknown. The goal is to reconstruct  $\Omega$ ,  $\mathbf{g}$ , and  $\mathbf{u}$  such that the  
 156 modeled velocity field is consistent with the observed data. To quantify the data misfit, we  
 157 define a discrepancy functional that measures the  $L^2$  distance between the measured velocity  
 158 and the simulated one as

$$159 \quad (3.1) \quad \mathbf{L}_{\text{data}}(\mathbf{u}) = \|\tilde{\mathbf{u}} - \mathbf{u}\|_{L^2(D)}^2,$$

160 The modeled velocity field is required to satisfy the steady incompressible Navier–Stokes  
 161 equations in the domain  $\Omega$ . Specifically, the velocity  $\mathbf{u}$  associated with a pressure  $p$  satisfies  
 162 the system

$$163 \quad (3.2) \quad \begin{cases} \mathbf{u} \cdot \nabla \mathbf{u} - \nabla p + \nu \Delta \mathbf{u} = \mathbf{0} & \text{in } \Omega, \\ \nabla \cdot \mathbf{u} = 0 & \text{in } \Omega, \\ \mathbf{u} = \mathbf{g} & \text{on } \Gamma^i, \\ (-p + \nu \nabla \mathbf{u}) \cdot \mathbf{n} = \mathbf{0} & \text{on } \Gamma^o, \\ \mathbf{u} = \mathbf{0} & \text{on } \Gamma^w. \end{cases}$$

164 for a domain  $\Omega \subset D$ , where  $\nu$  denotes the kinematic viscosity and  $\mathbf{n}$  is the outward normal  
 165 vector on the boundary. With this forward model (3.2), we could define a solution operator  
 166  $\mathcal{U}_\Omega$  that maps the inlet condition  $\mathbf{g}$  and the flow domain  $\Omega$  to the corresponding velocity field  
 167  $\mathbf{u}$ , that is,  $\mathcal{U}_\Omega : (\mathbf{g}, \Omega) \mapsto \mathbf{u}$  [32].

168 With these settings, we pose the inverse reconstruction problem as a PDE-constrained  
 169 optimization problem. Specifically, our objective is to recover the domain  $\Omega$  and the inlet  
 170 boundary condition  $\mathbf{g}$  by minimizing the discrepancy functional subject to the Navier–Stokes  
 171 equations, that is, we solve

$$172 \quad (3.3) \quad \min_{\mathbf{g}, \Omega} \mathbf{L}_{\text{data}}(\mathbf{u} = \mathcal{U}_\Omega[\mathbf{g}, \Omega]).$$

173 However, optimizing the flow domain  $\Omega$  is not straightforward and a specific representation  
 174 is required. Previous works have employed level-set methods [21] and phase-field approaches  
 175 [1]. In our approach, we model the domain using a quasi-conformal mapping, allowing for  
 176 both topological control of the evolving process and seamless integration into a network-based  
 177 optimization framework.

178 For the quasi-conformal mapping that will be introduced later, we need to use a reference  
 179 domain  $\Omega_0$  that has the same topology as the target domain  $\Omega$ . The actual flow domain  $\Omega$  is  
 180 then expressed by a quasi-conformal mapping  $f$  as

$$181 \quad (3.4) \quad \Omega = f(\Omega_0).$$

182 As a result, with  $\partial\Omega_0 = \Gamma_0^i \cup \Gamma_0^o \cup \Gamma_0^w$ , the boundaries of the deformed domain are expressed  
 183 in terms of the reference boundaries as

$$184 \quad (3.5) \quad \Gamma^i = f(\Gamma_0^i), \quad \Gamma^o = f(\Gamma_0^o), \quad \Gamma^w = f(\Gamma_0^w),$$

185 Consequently, we redefine the solution operator to account for the deformation, denoted by  
 186  $\mathcal{U}_f : (\mathbf{g}, f) \mapsto \mathbf{u}$ , mapping a boundary condition  $\mathbf{g}$  and a quasi-conformal mapping function  
 187  $f$  to a velocity field  $\mathbf{u}$  satisfying (3.2), (3.4) and (3.5). The data-fidelity-driven optimization  
 188 problem is then reformulated as

$$189 \quad (3.6) \quad \min_{\mathbf{g}, f} \mathbf{L}_{\text{data}}(\mathbf{u} = \mathcal{U}_f[\mathbf{g}, f]),$$

190 subject to the Navier–Stokes equations

$$191 \quad (3.7) \quad \left\{ \begin{array}{ll} \mathbf{u} \cdot \nabla \mathbf{u} - \nabla p + \nu \Delta \mathbf{u} = \mathbf{0} & \text{in } f(\Omega_0), \\ \nabla \cdot \mathbf{u} = 0 & \text{in } f(\Omega_0), \\ \mathbf{u} = \mathbf{g} & \text{on } f(\Gamma_0^i), \\ (-p + \nu \nabla \mathbf{u}) \cdot \mathbf{n} = \mathbf{0} & \text{on } f(\Gamma_0^o), \\ \mathbf{u} = \mathbf{0} & \text{on } f(\Gamma_0^w). \end{array} \right.$$

192 Despite this reformulation, the joint optimization over multiple coupled variables remains  
 193 challenging, especially given the dependency on a deforming spatial domain. To mitigate this  
 194 complexity, we adopt a Gauss–Seidel-style alternating optimization strategy. The optimization  
 195 is split into two subproblems. The first, termed the fluid subproblem, fixes the current domain  
 196 and solves for the velocity field  $\mathbf{u}$ , pressure  $p$ , and the inlet condition  $\mathbf{g}$ . The second, referred  
 197 to as the geometry subproblem, holds  $\mathbf{u}$  and  $\mathbf{g}$  fixed while optimizing the deformation mapping  
 198  $f$ , thus updating the flow domain.

199 **3.1. Fluid Problem.** For the fluid subproblem, we assume that the domain  $\Omega$  is fixed by  
 200 a known mapping  $f$ , i.e.,  $\Omega = f(\Omega_0)$ . With the geometry held constant, it becomes more  
 201 tractable to minimize the discrepancy between the observed velocity field  $\tilde{\mathbf{u}}$  and the modeled  
 202 solution  $\mathbf{u}$ , which satisfies the Navier–Stokes equations on the fixed domain. In this setting,  
 203 we redefine the solution operator  $\mathcal{U}_g : \mathbf{g} \mapsto \mathbf{u}$  that maps a given inlet boundary condition  
 204  $\mathbf{g}$  to the corresponding velocity field  $\mathbf{u}$ , while implicitly incorporating the fixed mapping  $f$ .  
 205 The resulting optimization problem becomes a partial minimization over  $\mathbf{g}$  with  $\mathbf{u} = \mathcal{U}_g[\mathbf{g}]$ ,  
 206 formulated as

$$207 \quad (3.8) \quad \min_{\mathbf{g}} \mathbf{L}_{\text{data}}(\mathbf{u} = \mathcal{U}_g[\mathbf{g}]).$$

208 Here, the deformation map  $f$  is fixed for this minimization problem, reducing the minimization  
 209 to a problem that only solves for the fluid dynamics.

210 **3.2. Geometry Problem.** To model the geometry subproblem, which aims to improve the  
 211 accuracy of domain estimation, it is essential to derive information that guides the optimiza-  
 212 tion of the deformation mapping. Note that although the observed velocity field is corrupted  
 213 by noise, it still reflects the flow in the true physical domain. Specifically, the reconstructed  
 214 flow field that satisfies (3.7) represents the template domain, as it is defined over the region  
 215  $f(\Omega_0)$ . Aligning this reconstructed field with the measured one thus drives the domain evo-  
 216 lution by deforming the template domain  $\Omega_0$ . Conceptually, such a registration map serves  
 217 as an approximation of the domain gradient, thereby guiding the deformation toward a more  
 218 accurate domain representation.

219 To this end, we seek a correction mapping  $c$  such that the composed domain  $c^{-1} \circ f(\Omega_0)$   
 220 more closely approximates the true flow domain. By extending the modeled velocity field  $\mathbf{u}$   
 221 with zero values in the region  $D \setminus \Omega$ , the refinement is achieved through a registration process  
 222 that aligns  $\mathbf{u}$  with the observed data  $\tilde{\mathbf{u}}$  through a registration procedure by minimizing

$$223 \quad (3.9) \quad \mathbf{L}_{\text{reg}} = \|\tilde{\mathbf{u}} - \mathbf{u} \circ c\|_{L^2(D)}^2,$$

224 where we observe that when the correction mapping  $c$  is an identity mapping, the optimization  
 225 functional  $\|\tilde{\mathbf{u}} - \mathbf{u} \circ c\|_{L^2(D)}$  reduces to the original data fidelity functional in  $\|\tilde{\mathbf{u}} - \mathbf{u}\|_{L^2(D)}$ .  
 226 This reflects the ideal scenario in which no further geometric correction is necessary, indicating  
 227 that the current domain accurately captures the true vascular region where the flow occurs.  
 228 The theoretical validation of this domain evolution formulation, that it admits a solution  
 229 minimizing the original objective (3.6) subject to the Navier–Stokes constraints in (3.7), will  
 230 be presented later in this section.

231 Both the correction mapping and the deformation mapping are modeled as quasi-conformal  
 232 mappings that are diffeomorphic. This design is essential to preserve the topology of the  
 233 reference domain, ensuring that  $\Omega = f(\Omega_0)$  retains the same topological structure as  $\Omega_0$ .  
 234 This property allows the model to incorporate morphological priors effectively. Furthermore,  
 235 this formulation is central to the theoretical analysis presented later. We give some detailed  
 236 explanations of the Beltrami mapping we shall use for the deformation in the following.

237 **Definition 3.1 (Quasi-conformal map).** *A quasi-conformal map is a map  $f : \mathbb{C} \rightarrow \mathbb{C}$  that*  
 238 *satisfies the Beltrami equation*

$$239 \quad (3.10) \quad \frac{\partial f}{\partial \bar{z}} = \mu(z) \frac{\partial f}{\partial z}$$

240 *for some complex-valued function called Beltrami coefficient  $\mu$  satisfying  $\|\mu\|_\infty < 1$  and such*  
 241 *that  $\frac{\partial f}{\partial \bar{z}}$  is non-vanishing almost everywhere. The complex partial derivatives are given by*

$$242 \quad \frac{\partial f}{\partial z} := \frac{1}{2} \left( \frac{\partial f}{\partial x} - i \frac{\partial f}{\partial y} \right) \quad \text{and} \quad \frac{\partial f}{\partial \bar{z}} := \frac{1}{2} \left( \frac{\partial f}{\partial x} + i \frac{\partial f}{\partial y} \right).$$

243

244 The Beltrami coefficient  $\mu$  quantifies the deviation from conformality. If  $\mu = 0$  at a point  $p$ ,  
 245 the map is conformal in a neighborhood around  $p$ , reducing (3.10) to the Cauchy–Riemann  
 246 equations. In this case,  $f$  near  $p$  can be written as

$$247 \quad f(z) = f(p) + f_z(p)(z + \mu(p)\bar{z}),$$

248 where  $f(p)$  is a translation and  $f_z(p)$  a conformal scaling. Thus,  $\mu$  captures the non-conformal  
 249 part of  $f$ , and  $f$  is conformal if and only if  $\mu \equiv 0$ .

250 Using a quasi-conformal mapping to represent the deformations of the domain, the geom-  
 251 etry subproblem can be finally formulated with respect to a correction mapping  $c$  as

$$252 \quad (3.11) \quad \min_c \mathbf{L}_{\text{reg}} = \|\tilde{\mathbf{u}} - \mathbf{u} \circ c\|_{L^2(D)}^2, \quad \text{s.t.} \quad \mu(c) < 1.$$

7

253 In our formulation, the quasi-conformal map  $c$  effectively serves as a domain gradient,  
 254 producing a smooth topology-preserving deformation that directly drives boundary evolution.  
 255 This ensures that the reconstructed geometry keeps the same topology as the template —  
 256 an important advantage in cardiovascular applications, where vessel networks have consistent  
 257 anatomical structures and the template provides a meaningful morphological prior. Even in  
 258 rare cases with atypical anatomy, a coarse template can still be extracted (e.g., from a rough  
 259 segmentation) and subsequently refined via flow-driven registration. Such topology preserva-  
 260 tion cannot be achieved by phase-based approaches, and level-set methods lack a theoretical  
 261 foundation for controlling it. Moreover, because the deformation is obtained by minimizing a  
 262 global registration energy over the entire flow field, the resulting geometry evolution is smooth,  
 263 stable, and globally coherent. This stands in contrast to the level-set approach, whose evolu-  
 264 tion is fundamentally local, driven by contour curvature and neighborhood features, making  
 265 it difficult to enforce global structure and long-range consistency during optimization.

266 **3.3. Subproblem Formulations Verification.** After decomposing the original problem into  
 267 the fluid and geometry subproblems, we now describe the alternating procedure used to iter-  
 268 atively solve these subproblems in order to approximate the solution of the original problem.

269 Let  $n$  denote the iteration number. Given a  $f_n$ , we first solve for iteration  $n$  by minimizing  
 270 the data discrepancy objective (3.8) subject to the Navier–Stokes constraints (3.7) over the  
 271 domain  $f_n(\Omega_0)$ , where the first mapping  $f_1$  is initialized (identity mapping in this work).  
 272 With the obtained  $u_n, p_n, g_n$  from the fluid subproblem (3.8), we then solve the geometry  
 273 subproblem 3.9 for the correction map to get its minimizer  $c_n$ . This map updates the domain  
 274 for the next iteration via  $f_{n+1} = c_n^{-1} \circ f_n$ . The process of alternatively solving  $u_n, p_n, g_n$  and  
 275  $c_n$  then repeats until convergence is met. We also refer to Figure 1 for the pipeline flow.

276 This alternating scheme captures the interplay between fluid physics and geometric de-  
 277 formation: The reconstructed velocity field informs the domain correction, which in turn  
 278 influences the next fluid solution. As the iterations progress, both the flow field  $u_n$  and the  
 279 mapping  $f_n$  are expected to converge to some limits that align with the measured data, the  
 280 governing physics, and anatomical constraints.

281 To support this methodology, we establish a theory that, under proper assumptions, the  
 282 converging sequence generated by this alternating procedure admits a partial and local mini-  
 283 mizer of the original problem. The full proof is provided in the appendix for completeness.

284 **Definition 3.2.** Let  $F(x, y)$  be a function defined on a product space  $\mathcal{X} \times \mathcal{Y}$ . The partial  
 285 minimization of  $F$  with respect to the variable  $y$ , for a fixed  $x \in \mathcal{X}$ , is defined as

$$286 \quad \min_{y \in \mathcal{Y}} F(x, y).$$

287 **Theorem 3.3.** Let  $\mathcal{H}(D)$  be the group of diffeomorphisms on a compact domain  $D$ . Let  $\Omega_0$   
 288 be a domain with smooth boundary and  $\tilde{u}$  measured data in  $D$ . Let  $\{(u_n, p_n, f_n, g_n, c_n)\}_{n=1}^\infty$  be a  
 289 sequence of solutions generated by an iterative inverse Navier–Stokes reconstruction algorithm,  
 290 which means

- 291 •  $f_1 \in \mathcal{H}(D)$  is given,
- 292 •  $(u_n, p_n, g_n)$  minimize  $\|\tilde{u} - u\|_{L^2(D)}^2$  with fixed  $f_n \in \mathcal{H}(D)$  for  $(u_n, p_n, g_n, f_n)$  satisfying  
 293 (3.7),

294 •  $f_{n+1} = c_n^{-1} \circ f_n$ , where  $c_n = \arg \min_{c \in \mathcal{H}(D)} \|\tilde{u} - u_n \circ c\|_{L^2(D)}$ .

295 Assume the following:

- 296 (1) The velocity fields converge:  $u_n \rightarrow u_*$  in  $L^\infty(D)$ ,  
 297 (2) The domain mappings converge:  $f_n \rightarrow f_*$  in  $L^\infty(D)$ ,  
 298 (3) In a neighborhood of  $f_*$  in  $L^\infty(D)$ , minimizing  $\|\tilde{u} - u\|_{L^2(D)}^2$  under (3.7) admits a  
 299 unique solution  $(u_n, p_n, g_n)$  and  $u \in W^{1,\infty}(D)$ . The solution operator  $\mathcal{U} : f \mapsto u$  is  
 300 continuous for the  $L^\infty(D)$  norm.  
 301 (4) For large  $n$ , the solution  $u_n$  is uniformly Lipschitz.

302 Then, the following conclusions hold:

- 303 (a) The correction mappings converge:  $c_n \rightarrow \mathbf{id}$  in  $L^\infty(D)$ ,  
 304 (b) The limit  $f_*$  is the partial and local minimizer under (3.7) for  $f \in \mathcal{H}(D)$ ,  
 305 (c) The limit  $u_*$  is the partial and local minimizer under (3.7) for  $f \in \mathcal{H}(D)$ ,

306 Note that assumptions (1)-(2) stem from empirical observations on our numerical scheme.  
 307 For all cases presented in this paper, the iterative scheme converges in a manner consistent  
 308 with these assumptions. The uniqueness and the regularity of the solution to the Navier-Stokes  
 309 equations (i.e., equation (3.7) with an appropriate choice of boundary data  $\mathbf{g}$ ) are established  
 310 in [33, Theorem 3.7.3] and [19, Theorem 7.1]. Therefore, given proper smoothness of  $\Omega_0$  and  
 311  $f_n$ , assuming the uniqueness of the associated minimization problem and the uniform Lipschitz  
 312 continuity of the solution sequence is reasonable and consistent with the well-posedness of the  
 313 PDE constraint.

314 **4. Numerical Method.** In the following sections, we provide specific implementation de-  
 315 tails, including the point sampling strategy within the region and along the boundaries defined  
 316 by the template mask, the formulation of the fluid subproblem, and the geometry modeling.  
 317 Finally, we present the complete algorithm.

318 **4.1. Mask Sampling for Flow Domain.** Since a binary mask is used to represent the flow  
 319 region where blood is expected to circulate, special consideration is required when sampling  
 320 points for PINN training. We now describe our sampling strategy for the fluid domain  $\Omega$  and  
 321 its boundary  $\partial\Omega$  at iteration  $n$ . Denoting the deformation mapping at this iteration by  $f$ , the  
 322 domain is given by  $\Omega = f(\Omega_0)$ . Similarly, as in (3.5), the boundary components satisfy

$$323 \quad (4.1) \quad \Gamma^i = f(\Gamma_0^i), \quad \Gamma^o = f(\Gamma_0^o), \quad \Gamma^w = f(\Gamma_0^w),$$

324 The input data consists of a velocity image  $\tilde{\mathbf{u}}$  that is defined on  $[0, \varepsilon H] \times [0, \varepsilon K]$ , where  $\varepsilon$  is  
 325 determined by the image resolution. Each pixel point is positioned at coordinates

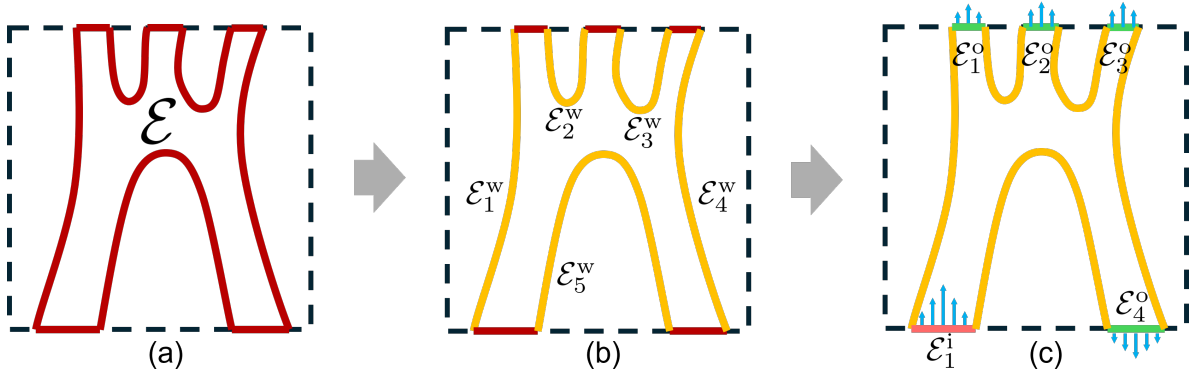
$$326 \quad (4.2) \quad \mathcal{D} := \{(\varepsilon/2 + h \cdot \varepsilon, \varepsilon/2 + k \cdot \varepsilon) : \text{where } h = 0, \dots, H-1 \text{ and } k = 0, \dots, K-1\}.$$

327 Note that although the value for  $\tilde{\mathbf{u}}$  is only explicitly defined on the grid points in  $\mathcal{D}$ , its values  
 328 at other locations can be obtained through interpolation methods; in this work, bilinear  
 329 interpolation is used for this purpose.

330 Let us also denote the mask that is used to represent the domain  $\Omega$  as  $M_\Omega$  defined on  
 331 pixel points  $\mathcal{D}$  as

$$332 \quad (4.3) \quad M_\Omega = \begin{cases} 1 & \text{for } \mathbf{x} \in \mathcal{W}, \\ 0 & \text{for } \mathbf{x} \in \mathcal{D} \setminus \mathcal{W}. \end{cases}$$





**Figure 2.** Illustration of the steps used to divide the entire boundary into wall, outlet, and inlet boundaries. Red indicates the full boundary or the unclassified segments; yellow denotes wall boundaries; green represents outlets; and pink indicates inlets.

348 where each of the point sets is 4-connected, and the sets are mutually disjoint. To determine  
 349 whether a given point collection  $\mathcal{E}_z^*$  (where  $*$   $\in$   $\{i, o\}$ ) corresponds to an inlet or outlet), we  
 350 first compute the average velocity over the points in the set as follows:

351 (4.5) 
$$\bar{\mathbf{u}}^* = \frac{1}{|\mathcal{E}^*|} \sum_{\mathbf{x} \in \mathcal{E}^*} \mathbf{u}(\mathbf{x}).$$

352 where  $|\mathcal{E}_z^*|$  represent the number of points in  $\mathcal{E}_z^*$ . Then, as in Figure 2.(c), if  $\bar{\mathbf{u}}^*$  points inward  
 353 to the image domain, which indicates an inflow pattern,  $\mathcal{E}^*$  should be an inlet boundary.  
 354 Otherwise, the average velocity points outward from the image domain, indicating an outflow  
 355 pattern, and thus the collection corresponds to an outlet.

356 Thus, the pixel points for the interior region of  $\Omega$  is as

357 
$$\mathcal{G} = \mathcal{W} - \mathcal{E}.$$

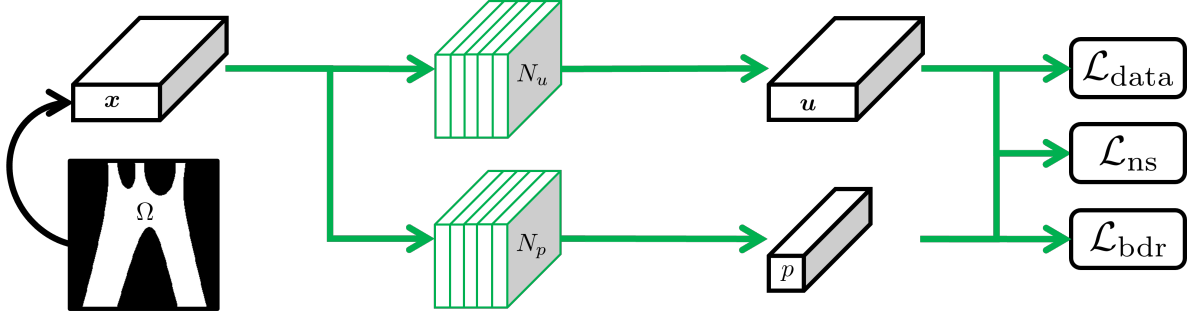
358 and with  $\mathcal{G}$  and  $\mathcal{E}$  in hand, we can then do sampling for the interior domain  $\Omega \setminus \partial\Omega$  and the  
 359 boundary  $\Gamma$ .

360 To sample a new point in the interior domain  $\Omega \setminus \partial\Omega$ , we could first randomly choose one  
 361 point  $\mathbf{p}$  from  $\mathcal{G}$ . Then, the sampled point is as

362 (4.6) 
$$\mathbf{x}^{\mathcal{G}} = \mathbf{p} + [l_1, l_2]',$$

363 where  $l_1$  and  $l_2$  are randomly sampled from  $(-\epsilon, \epsilon)$ .

364 To sample points along a segment of  $\mathcal{E}_z^i$ ,  $\mathcal{E}_z^o$ , and  $\mathcal{E}_z^w$ , we apply piecewise linear inter-  
 365 polation. Specifically, let  $\mathcal{E}^*$  denote one such segment consisting of an ordered set of points  
 366  $\{\mathbf{p}_0, \mathbf{p}_1, \dots, \mathbf{p}_{N_*}\}$ , where each consecutive pair  $(\mathbf{p}_j, \mathbf{p}_{j+1})$  defines a straight-line segment. Then,  
 367 we first generate a random value  $l \in [0, L]$ , where  $L$  is the total length of the boundary. We  
 368 then determine which segment the sampled point will lie in by accumulating in each segment's



**Figure 3.** Architecture of the PINN for fluid flow prediction, featuring two separate networks dedicated to velocity and pressure estimation.

369 length  $S_r$  until  $L_j = \sum_{r=1}^j S_r \geq l$  is met, which means the sampled points within segment  $j$   
 370 as  $(p_{j-1}, p_j)$ . We finally have the sampled point on the boundary as

$$371 \quad (4.7) \quad \mathbf{x}^e = \frac{L_j - l}{L_j - L_{j-1}} \mathbf{p}_{j-1} + \frac{l - L_{j-1}}{L_j - L_{j-1}} \mathbf{p}_j.$$

372 **4.2. Fluid Subproblem.** We describe some details for optimizing the fluid subproblems  
 373 (3.8) for each iteration  $n$  through Physics-Informed Neural Networks. For convenience, we  
 374 denote the current spatial domain as  $\Omega$ , omitting the iteration index. This represents  $\Omega_n$  at  
 375 iteration  $n$ . We first randomly sample  $M^g$  points from the inner domain as  $\mathbf{x}_m^g \in \Omega \setminus \partial\Omega$   
 376 according to (4.6) for  $m = 1, \dots, M^g$ . Regarding the boundaries, we randomly sample the  
 377 coordinates  $M^i$ ,  $M^o$  and  $M^w$  at the inlet, outlet and wall boundary as  $\mathbf{x}_m^i \in \Gamma^i$ ,  $\mathbf{x}_m^o \in \Gamma^o$   
 378 and  $\mathbf{x}_m^w \in \Gamma^w$  according to (4.7), respectively. We then introduce two networks as given  
 379 in Figure 3: one for the velocity computation, named  $\mathcal{N}_u$ , and another for the pressure  
 380 computation, named  $\mathcal{N}_p$ . We use  $\theta_u$  and  $\theta_p$  the trainable parameters for these two neural  
 381 networks. The two networks map any point  $\mathbf{x} \in \Omega$  to the values of the velocity and pressure  
 382 at this point, respectively. Omitting the subscripts of the sampled points, the flow velocity  
 383 and pressure for the point  $\mathbf{x}^g$  can be calculated as

$$384 \quad (4.8) \quad \begin{aligned} \mathbf{u}^g(\mathbf{x}^g; \theta_u) &= \mathcal{N}_u(\mathbf{x}^g; \theta_u), \\ p^g(\mathbf{x}^g; \theta_p) &= \mathcal{N}_p(\mathbf{x}^g; \theta_p), \end{aligned}$$

385 where for the boundaries, the notations are changed correspondingly as  $\mathbf{u}^i$ ,  $\mathbf{u}^o$ ,  $\mathbf{u}^w$  and  $p^o$ .  
 386 Also, let's define the discretized norm used throughout. Let  $U$  denote one of the relevant  
 387 domains:  $\Omega$ ,  $\Gamma^i$ ,  $\Gamma^o$ , or  $\Gamma^w$ . For each domain  $U$ , let  $\{\mathbf{x}_m\}_{m=1}^M$  represent the corresponding  
 388 set of sampled points (indexed appropriately). Given a function  $F : U \rightarrow \mathbb{R}$ , we define the  
 389 discrete  $L^2$  norm as

$$390 \quad (4.9) \quad \|F\|_{L^2(U;M)} := \left( \frac{1}{M} \sum_{m=1}^M |F(\mathbf{x}_m)|^2 \right)^{1/2} \quad \text{where } \mathbf{x}_m \sim U.$$

391 To train the networks, in addition to the measured data, the governing partial differential  
 392 equations should also be incorporated into the loss function through their residuals to make  
 393 the neural networks physics-informed. Since the networks take spatial coordinates as input  
 394 and predict physical quantities such as velocity and pressure, automatic differentiation (e.g.,  
 395 in PyTorch or TensorFlow) can be used to compute the derivatives required to evaluate these  
 396 residuals. We then present the details of these residuals, which together constitute the loss  
 397 function.

398 As the modeled velocity is extended by zero value over  $D \setminus \Omega$ , the data fitting term is thus  
 399 only defined in the domain  $\Omega$  as

$$400 \quad (4.10) \quad \mathcal{L}_{\text{data}}(\boldsymbol{\theta}_u, \boldsymbol{\theta}_p) = \|\tilde{\mathbf{u}} - \mathbf{u}^g\|_{L^2(\Omega; M^g)}^2.$$

401 To make the process physics-informed, we rewrite the Navier-Stokes equation into residual  
 402 forms

$$403 \quad (4.11) \quad \mathcal{L}_{\text{ns}}(\boldsymbol{\theta}_u, \boldsymbol{\theta}_p) = \|\mathbf{u}^g \cdot \nabla \mathbf{u}^g - \nabla p^g + \nu \Delta \mathbf{u}^g\|_{L^2(\Omega; M^g)}^2 \\ + \|\nabla \cdot \mathbf{u}^g\|_{L^2(\Omega; M^g)}^2,$$

404 and its associated boundary conditions into residual form as

$$405 \quad (4.12) \quad \mathcal{L}_{\text{bdr}}(\boldsymbol{\theta}_u, \boldsymbol{\theta}_p) = \|\mathbf{u}^i - \mathbf{g}^i\|_{L^2(\Gamma^i; M^i)}^2 + \|(-p^o + \nu \nabla \mathbf{u}^o) \cdot \mathbf{n}\|_{L^2(\Gamma^o; M^o)}^2 + \|\mathbf{u}^w\|_{L^2(\Gamma^w; M^w)}^2.$$

406 We also assume that the inlet velocity boundary follows a parabolic profile. Therefore, the  
 407 inlet velocity prediction should be regularized to be a parabolic velocity profile whose average  
 408 velocity matches the average noisy velocity over  $\Gamma^i$ , given by

$$409 \quad \mathbf{g}^r(\mathbf{x}) = \frac{3}{2} \frac{\int_{\Gamma^i} \tilde{\mathbf{u}}}{|\Gamma^i|} \left(1 - \frac{|\mathbf{q} - \mathbf{x}|^2}{(|\Gamma^i|/2)^2}\right),$$

410 where  $\mathbf{q}$  is the centroid of  $\Gamma^i$  and  $|\Gamma^i|$  represents the length of it. This leads to an additional  
 411 regularization for the inlet velocity as

$$412 \quad (4.13) \quad \mathcal{L}_{\text{in}} = \|\mathbf{u}^i - \mathbf{g}^r\|_{L^2(\Gamma^i; M^i)}^2,$$

413 Integrating the losses from both the flow and deformation problems with proper boundary  
 414 losses, we obtain

$$415 \quad (4.14) \quad \mathcal{L}_{\text{fluid}}(\boldsymbol{\theta}_u, \boldsymbol{\theta}_p) = \alpha_{\text{data}} \mathcal{L}_{\text{data}} + \alpha_{\text{ns}} \mathcal{L}_{\text{ns}} + \alpha_{\text{bdr}} \mathcal{L}_{\text{bdr}} + \alpha_{\text{in}} \mathcal{L}_{\text{in}},$$

416 where  $\alpha_{\text{data}}$ ,  $\alpha_{\text{ns}}$ ,  $\alpha_{\text{bdr}}$  and  $\alpha_{\text{in}}$  are the weights for reconstruction error, the Navier-Stokes  
 417 equations, the boundary condition and inlet velocity regularization respectively. Then, the  
 418 fluid optimization problem is modeled as follows,

$$419 \quad (4.15) \quad \min_{\boldsymbol{\theta}_u, \boldsymbol{\theta}_p} \mathcal{L}_{\text{fluid}}(\boldsymbol{\theta}_u, \boldsymbol{\theta}_p).$$

420 To reduce the loss values, gradient descent methods are used to update  $\boldsymbol{\theta}_u$  and  $\boldsymbol{\theta}_p$  with  
 421 the help of two distinct optimizers by

$$422 \quad (4.16) \quad \boldsymbol{\theta}_u \leftarrow \boldsymbol{\theta}_u - \tau_u \nabla_{\boldsymbol{\theta}_u} \mathcal{L}_{\text{fluid}}(\boldsymbol{\theta}_u, \boldsymbol{\theta}_p), \\ \boldsymbol{\theta}_p \leftarrow \boldsymbol{\theta}_p - \tau_p \nabla_{\boldsymbol{\theta}_p} \mathcal{L}_{\text{fluid}}(\boldsymbol{\theta}_u, \boldsymbol{\theta}_p).$$

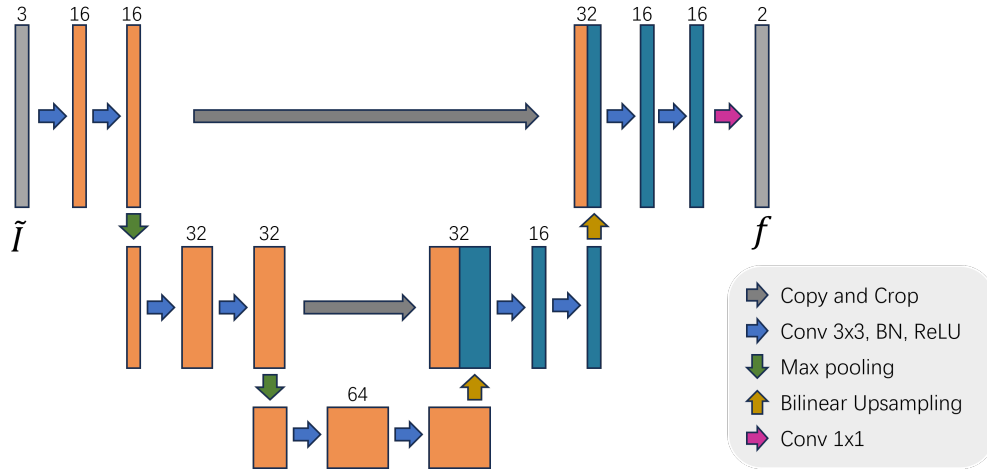


Figure 4. Architecture of the UNet used for quasi-conformal mapping estimation.

423 where  $\nabla_{\theta_u}$  and  $\nabla_{\theta_p}$  are the gradients with respect to  $\theta_u$  and  $\theta_p$ , which are computed by  
 424 autograd within *PyTorch*. Here,  $\tau_u$  and  $\tau_p$  represent adaptive step sizes determined by  
 425 various methods [5, 20], with step sizes computed according to the approach outlined in [20].

426 After sufficiently training the neural networks  $\mathcal{N}_u$  and  $\mathcal{N}_p$  in iteration  $n$  by minimizing  
 427 (4.14) with  $\Omega = \Omega_n$  in (4.11)–(4.13), we sample the reconstructed flow image over the entire  
 428 image grid  $\mathcal{D}$  (defined in (4.2) for use in the geometry subproblem to be detailed in the next  
 429 subsection. The sampled velocity field  $\hat{\mathbf{u}}$  is defined as:

$$430 \quad (4.17) \quad \hat{\mathbf{u}}(\mathbf{x}) = \begin{cases} \mathbf{u}^g(\mathbf{x}), & \text{if } \mathbf{x} \in \mathcal{G}, \\ \mathbf{u}^i(\mathbf{x}), & \text{if } \mathbf{x} \in \mathcal{E}^i, \\ \mathbf{u}^o(\mathbf{x}), & \text{if } \mathbf{x} \in \mathcal{E}^o, \\ \mathbf{u}^w(\mathbf{x}), & \text{if } \mathbf{x} \in \mathcal{E}^w, \\ 0, & \text{otherwise.} \end{cases}$$

431 Here,  $\mathcal{G}$ ,  $\mathcal{E}^i$ ,  $\mathcal{E}^o$ , and  $\mathcal{E}^w$  denote the sets of points in the discretized image grid corresponding  
 432 to  $\Omega$ ,  $\Gamma^i$ ,  $\Gamma^o$ , and  $\Gamma^w$  as described in Section 4.1, respectively.

433 **4.3. Geometry Subproblem.** We now address the geometry subproblem for each iteration  
 434  $n$ . We first obtain the value of solved  $\hat{\mathbf{u}}$  from the fluid subproblem. Then, we also define the  
 435 norm over the discretized grid  $\mathcal{D}$  for a function  $F : \mathcal{D} \rightarrow \mathbb{R}$  as

$$436 \quad (4.18) \quad \|F\|_{L^2(\mathcal{D})} = \left( \frac{1}{HK} \sum_{\mathbf{x} \in \mathcal{D}} |F(\mathbf{x})|^2 \right)^{1/2},$$

437 As discussed in the previous section, we need to register the image  $\hat{\mathbf{u}}$  obtained from the  
 438 solution of the fluid subproblem (4.15) to the captured noisy image  $\tilde{\mathbf{u}}$  to produce a mapping  
 439  $\mathbf{c}$ , which serves as an approximation of the domain gradient. This mapping is generated by a

440 neural network

$$441 \quad (4.19) \quad c = \mathcal{N}_c(\hat{\mathbf{u}}; \boldsymbol{\theta}_c)$$

442 where  $\boldsymbol{\theta}_c$  is the trainable weights of the network. Following the approach of [46, 47], we  
 443 use a U-Net architecture for  $\mathcal{N}_c$ , as illustrated in Figure 4, to estimate the quasi-conformal  
 444 mapping. To ensure accurate registration, the network is trained by minimizing the following  
 445 registration loss.

$$446 \quad (4.20) \quad \mathcal{L}_{\text{reg}}(\boldsymbol{\theta}_c) = \|\tilde{\mathbf{u}} - \hat{\mathbf{u}} \circ c\|_{L^2(\mathcal{D})}^2,$$

447 For  $c$  to be a quasi-conformal mapping, its associated Beltrami coefficient must be strictly less  
 448 than 1 according to Definition 3.1. In addition, to promote the mapping  $c$  to be diffeomorphic,  
 449 we apply Laplacian regularization to encourage smoothness and control geometric distortion.  
 450 Thereby, the regularization terms are introduced as

$$451 \quad (4.21) \quad \begin{aligned} \mathcal{L}_{\text{bc}}(\boldsymbol{\theta}_c) &= \|e^{\mu(c)-1}\|_{L^2(\mathcal{D})}^2, \\ \mathcal{L}_{\text{lap}}(\boldsymbol{\theta}_c) &= \|\Delta c\|_{L^2(\mathcal{D})}^2, \end{aligned}$$

452 where  $\mu(c)$  represents the Beltrami coefficient of the mapping  $c$ , computed through a finite  
 453 difference implementation of Equation (3.10). The term  $\Delta c$  denotes the Laplacian of the map-  
 454 ping  $c$ . Together, these regularizations ensure that  $c$  adheres to quasi-conformal constraints  
 455 while maintaining desirable geometric properties, such as continuity and smoothness.

456 The overall objective function for the geometry subproblems (3.9) is then replaced by

$$457 \quad (4.22) \quad \mathcal{L}_{\text{geo}}(\boldsymbol{\theta}_c) = \alpha_{\text{reg}}\mathcal{L}_{\text{reg}} + \alpha_{\text{bc}}\mathcal{L}_{\text{bc}} + \alpha_{\text{lap}}\mathcal{L}_{\text{lap}},$$

458 where  $\alpha_{\text{reg}}$ ,  $\alpha_{\text{bc}}$ , and  $\alpha_{\text{lap}}$  are the weighting parameters for the registration error, Beltrami  
 459 regularization, and Laplacian regularization, respectively. Then, the geometry subproblem is  
 460 reformulated as

$$461 \quad (4.23) \quad \min_{\boldsymbol{\theta}_c} \mathcal{L}_{\text{geo}}(\boldsymbol{\theta}_c).$$

462 To reduce the loss values, gradient descent methods are used to update  $\boldsymbol{\theta}_c$  by

$$463 \quad (4.24) \quad \boldsymbol{\theta}_c \leftarrow \boldsymbol{\theta}_c - \tau_c \nabla_{\boldsymbol{\theta}_c} \mathcal{L}_{\text{geo}}(\boldsymbol{\theta}_c),$$

464 where  $\nabla_{\boldsymbol{\theta}_c}$  is the gradient with respect to  $\boldsymbol{\theta}_c$ ,  $\tau_c$  represents adaptive step sizes as in [5]. After  
 465 sufficient training, the produced mapping is used to warp the current mask, thereby refining  
 466 the vessel region. This updated domain is then used in the next iteration of fluid subproblem  
 467 training.

468 **4.4. Overall Algorithm.** The proposed algorithm iteratively alternates between solving  
 469 the flow subproblem (4.15) and the geometry subproblem (4.23) to achieve accurate blood  
 470 flow imaging. As a result, the overall optimization problem (3.6), subject to the constraint in  
 471 (3.7), is approximately solved through this alternating scheme between the fluid subproblem  
 472 (4.15) and the geometry subproblem (4.23).

473 We begin with the fluid subproblem using the domain mask  $M_{\Omega_n}$ , which represents the  
474 mask at iteration  $n$ , as defined in (4.3). The initial mask  $M_{\Omega_1}$ , provided at the beginning,  
475 serves as both a topological prior and a coarse approximation of the spatial domain in which  
476 blood flow is expected to occur. Based on the current mask  $M_{\Omega_n}$ , spatial points are sampled  
477 and used as input to the PINN models  $\mathcal{N}_u$  and  $\mathcal{N}_p$  in (4.8) used for (4.15). These two  
478 networks are then trained on the sampled points by minimizing the fluid loss function defined  
479 in (4.14), using the gradient descent described in (4.16). Once sufficiently trained, the models  
480 produce a velocity field that adheres to the governing physical laws (3.7) while fitting the  
481 noisy observational data  $\tilde{u}$ .

482 After training, we evaluate the predicted velocity at the points of the image grid  $\mathcal{D}$  (as  
483 defined in (4.2)), which produces a reconstructed flow image  $\hat{u}_n$ , which corresponds to the  
484 indexed sampled velocity image at iteration  $n$ , denoted by  $\hat{\mathbf{u}}$  in (4.17). To correct for potential  
485 misalignments between the solved domain of this iteration and the actual vessel geometry, we  
486 estimate a quasi-conformal registration mapping that aligns  $\hat{u}_n$  with the observed flow  $\tilde{u}$ .  
487 This is achieved by training the mapping network  $\mathcal{N}_c$  to minimize the registration loss (4.22),  
488 using the gradient descent scheme in (4.24). After sufficient training, the network outputs a  
489 registration mapping  $c_n$ , which is used to update the vessel mask by warping the previous  
490 domain:  $M_{\Omega_{n+1}} = M_{\Omega_n} \circ c_n$ . As shown in Remark 1, this is equivalent to  $M_{\Omega_{n+1}} = M_{c_n^{-1}(\Omega_n)}$ .  
491 The updated mask  $M_{\Omega_{n+1}}$  then replaces the previous one and serves as the domain mask for  
492 the next iteration, as illustrated in Figure 1, enabling a new round of point sampling for the  
493 fluid subproblem training.

494 This alternation between the PINN-based flow reconstruction problem (4.15) and domain  
495 refinement through the quasi-conformal registration problem (4.23) continues iteratively. The  
496 algorithm terminates when either a convergence criterion is satisfied or the maximum number  
497 of iterations is reached. The overall algorithm is summarized in Algorithm 4.1 for a clear  
498 presentation.

499 **Remark 1.** *The two formulations*

500 (4.25) 
$$M_{\Omega_{n+1}} = M_{c_n^{-1}(\Omega_n)} \quad \text{and} \quad M_{\Omega_{n+1}} = M_{\Omega_n} \circ c_n$$

501 *are equivalent under the interpretation of  $M_{\Omega}$  as the characteristic function of the domain  $\Omega$ .*  
502 *This equivalence is seen by evaluating the composition:*

503 (4.26) 
$$M_{\Omega_n} \circ c_n(x) = 1 \quad \iff \quad c_n(x) \in \Omega_n \quad \iff \quad x \in c_n^{-1}(\Omega_n).$$

504 *Therefore,  $M_{\Omega_n} \circ c_n$  is the characteristic function of  $c_n^{-1}(\Omega_n)$ , which justifies the update*  
505  *$M_{\Omega_{n+1}} = M_{\Omega_n} \circ c_n$  as a computationally convenient form to generate the warped mask.*

## 506 **5. Experiment.**

### 507 **5.1. Implementation Details.**

508 **Data Synthesis** Synthetic data were generated using MATLAB FEATool with the inlet  
509 velocity profile

510 (5.1) 
$$\mathbf{g}(\mathbf{x}) = v \left( 1 - \frac{|\mathbf{q} - \mathbf{x}|^2}{(|\Gamma^i|/2)^2} \right),$$

---

**Algorithm 4.1** The main algorithm

---

**Require:** Measured flow image  $\tilde{\mathbf{u}}$ , reference mask  $M_{\Omega_0}$ ;

**Ensure:** Reconstructed flow image  $\mathbf{u}$ , reconstructed mask  $M_{\Omega}$ ;

```
1:  $M_{\Omega_1} \leftarrow M_{\Omega_0}$ ;  
2:  $\boldsymbol{\theta}_u, \boldsymbol{\theta}_p, \boldsymbol{\theta}_c \leftarrow$  random initialization;  
3: for  $n = 1 : N_{\text{MAX}}$  do  
4:   Sample points using  $M_{\Omega_n}$  as described in Section 4.1;  
5:    $\boldsymbol{\theta}_u, \boldsymbol{\theta}_p \leftarrow$  gradient descent as (4.16) with the sampled points;  
6:   for each  $\mathbf{x} \in \mathcal{D}$  (defined in (4.2)) do  
7:      $\hat{\mathbf{u}}(\mathbf{x}) \leftarrow \mathcal{N}_u(\mathbf{x}; \boldsymbol{\theta}_u)$  ;  
8:   end for  
9:    $\hat{\mathbf{u}}_n \leftarrow \hat{\mathbf{u}}$ ;  
10:  if  $\|\tilde{\mathbf{u}} - \hat{\mathbf{u}}_n\|_{L^2(\mathcal{D})}^2$  converge then  
11:     $\mathbf{u} \leftarrow \hat{\mathbf{u}}_n$ ,  $M_{\Omega} \leftarrow M_{\Omega_n}$ ;  
12:    Finish and output;  
13:  end if  
14:   $\boldsymbol{\theta}_c \leftarrow$  gradient descent as (4.24) with  $\hat{\mathbf{u}}$ ;  
15:   $c_n \leftarrow \mathcal{N}_c(\hat{\mathbf{u}}_n; \boldsymbol{\theta}_c)$ ;  
16:   $M_{\Omega_{n+1}} \leftarrow M_{\Omega_n} \circ c_n$ ;  
17: end for
```

---

511 where  $v$  is a velocity parameter chosen for a specific experiment,  $\mathbf{q}$  denotes the centroid of the  
512 inlet boundary  $\Gamma^i$ , and  $|\Gamma^i|$  represents its length (in 2D). All synthesized images were of size  
513  $256 \times 256$ . The spatial resolution for each experiment is reported individually.

514 **Noise Model** To simulate the measurement noise in MRI flow imaging, we adopted a  
515 signal-dependent noise model combining Gaussian components

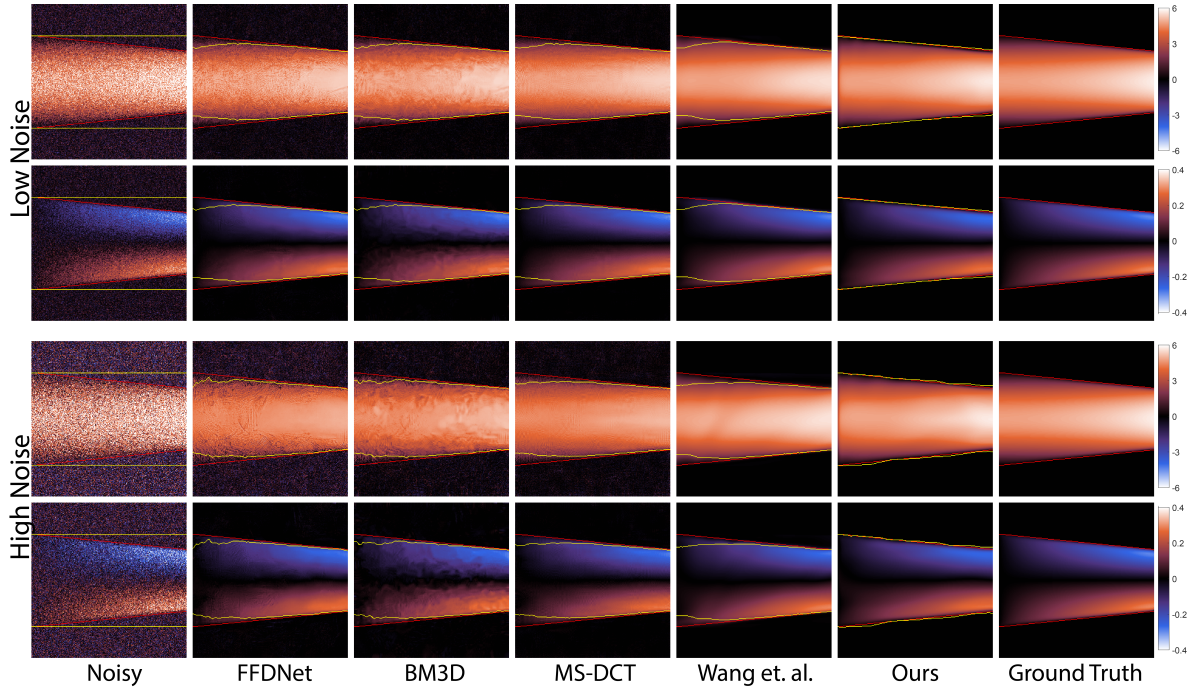
$$516 \quad (5.2) \quad \tilde{\mathbf{u}} = \mathbf{u} + \sqrt{|\mathbf{u}|} \cdot \mathbf{n}_1 + \mathbf{n}_2,$$

517 where  $\mathbf{n}_1 \sim \mathcal{N}(0, \sigma_1^2)$  and  $\mathbf{n}_2 \sim \mathcal{N}(0, \sigma_2^2)$  are two independent Gaussian noise terms.

518 **Network Architecture** The PINN model consists of two separate MLPs for velocity and  
519 pressure. The velocity network has 8 layers with 40 neurons per layer, while the pressure net-  
520 work also has 8 layers with 20 neurons per layer. Increasing the number of layers can improve  
521 expressiveness, but can lead to a higher computational cost without significant performance  
522 gains [44]. The U-Net architecture used to generate the correction mapping is shown in Fig-  
523 ure 4 and has been demonstrated to be effective for producing high-quality quasi-conformal  
524 mappings in previous works [43, 45].

525 **Error Metric** To evaluate the performance of our method, we use several metrics, includ-  
526 ing Mean Squared Error (MSE), Relative Error (RE), Peak Signal-to-Noise Ratio (PSNR),  
527 Structural Similarity Index (SSIM), Dice coefficient, and 95th percentile Hausdorff distance  
528 (HD95). Detailed definitions of these metrics are provided in Appendix B.

529 **Computational Platform** Our methods were implemented in all experiments using the  
530 *Python* programming language in combination with the *PyTorch* framework. The experiments  
531 were conducted on a Windows 11 PC equipped with an Nvidia 4060 Ti GPU and a 13th Gen



**Figure 5.** Qualitative comparison of flow and region reconstruction. Red contours show ground truth; yellow contours indicate predicted domains. Top two rows: low noise group; bottom two: high noise group. Each pair shows  $x$ - (top) and  $y$ -direction (bottom) velocity. Yellow contour in "Noisy" column marks the initialization.

532 Intel(R) Core(TM) i7-13700KF 3.40 GHz CPU.

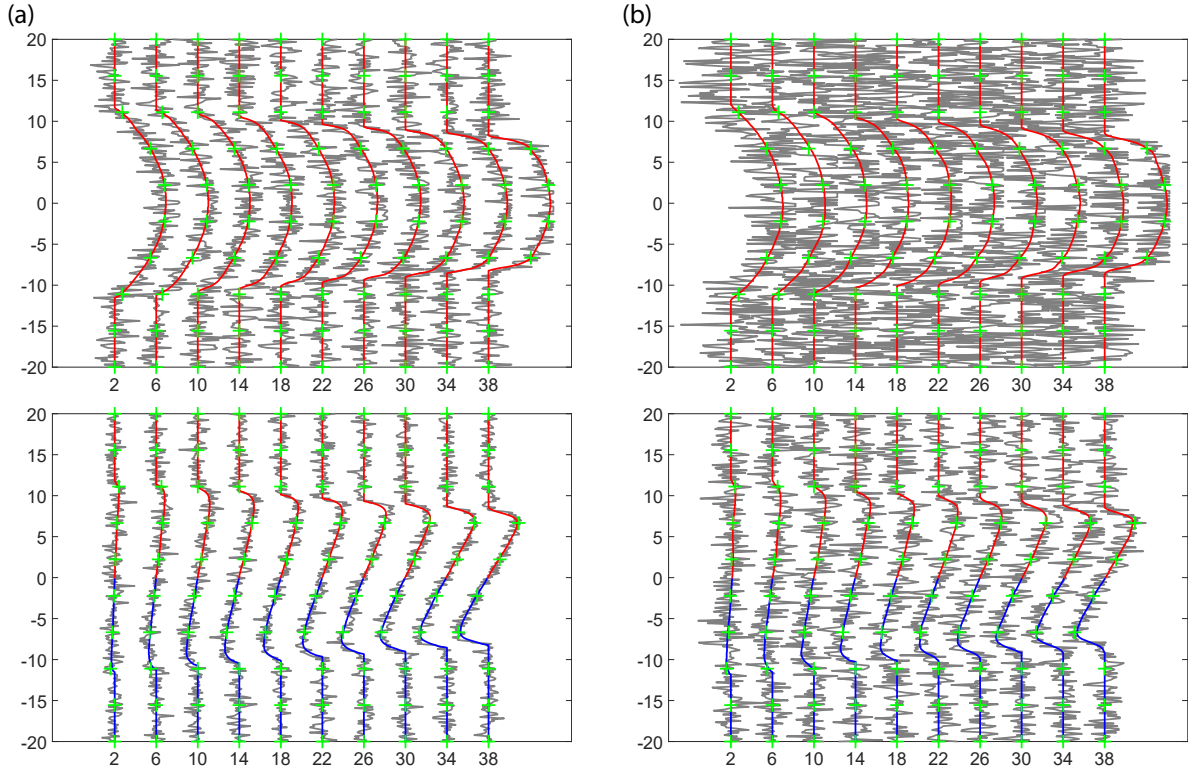
533 **Parameters** We set the weights of the composite loss function as follows if not specified:  
 534 for fluid subproblem,  $\alpha_{\text{data}} = 1.0$ ,  $\alpha_{\text{pde}} = 1.0 \times 10^{-3}$ ,  $\alpha_{\text{bd}} = 1.0$ ,  $\alpha_{\text{in}} = 0.1$ ; for geometry  
 535 subproblem,  $\alpha_{\text{reg}} = 1.0$ ,  $\alpha_{\text{bc}} = 1.0 \times 10^1$ ,  $\alpha_{\text{lap}} = 1$ .

536 **5.2. Flow Image over Converging Channel.** To validate the proposed method, we con-  
 537 ducted experiments on synthetic flow images within a converging channel. The data consists  
 538 of a two-channel image of size  $256 \times 256$ , representing the vertical and horizontal components  
 539 of the velocity field over a physical domain measuring  $40 \text{ cm} \times 40 \text{ cm}$ . In the simulation  
 540 setup, the inlet boundary velocity follows a parabolic profile with a maximum velocity of  
 541  $5 \text{ cm/s}$ . We assume a Newtonian fluid with dynamic viscosity  $\mu = 0.035 \text{ g} \cdot \text{cm}^{-1} \cdot \text{s}^{-1}$  and  
 542 density  $\rho = 1.05 \text{ g} \cdot \text{cm}^{-3}$ , corresponding to a Reynolds number of 150 based on a charac-  
 543 teristic length of  $5 \text{ cm}$  and a characteristic velocity of  $1 \text{ cm/s}$ . Either low or high levels of  
 544 Gaussian noise were added to the synthesized images. For low noise, the  $x$ -direction flow had  
 545 zero-mean noise with standard deviation 1, and the  $y$ -direction flow had zero-mean noise with  
 546 standard deviation 0.01. For high noise, the  $x$ -direction flow had a standard deviation of 2,  
 547 and the  $y$ -direction flow had a standard deviation of 0.02. The training was conducted for 240  
 548 iterations, alternating between 40 iterations of the fluid subproblem and 40 iterations of the  
 549 geometry subproblem. Each fluid subproblem involves 50,000 collocation points. The entire

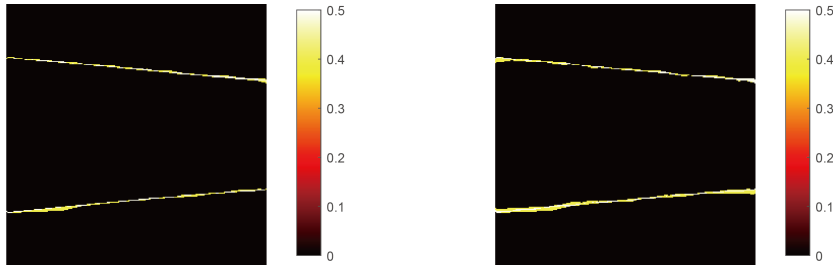
		Flow Image Reconstruction				Fluid Domain Segmentation		
		RE	MSE	PSNR	SSIM	MSE	Dice	HD95
Low	Noise	0.4974	0.4742	23.119	0.1962	0.0961	0.9119	24.739
	FFDNet	0.0827	0.0973	32.935	0.7452	0.0454	0.9522	16.454
	BM3D	0.0348	0.0483	31.276	0.8870	0.0457	0.9518	16.342
	MC-DCT	0.0411	<b>0.0383</b>	34.523	0.9172	0.0476	0.9498	16.616
	Wang <i>et al.</i>	0.0648	0.0486	34.921	<b>0.9855</b>	0.0407	0.9573	16.000
	Ours	<b>0.0124</b>	0.0414	<b>35.298</b>	0.9847	<b>0.0067</b>	<b>0.9933</b>	<b>4.000</b>
High	Noise	1.9621	1.7588	17.304	0.0645	0.0961	0.9119	24.739
	FFDNet	0.1656	0.3241	29.386	0.6564	0.0521	0.9358	19.417
	BM3D	0.1279	0.1012	32.130	0.7733	0.0529	0.9350	20.105
	MC-DCT	0.0891	0.0727	31.283	0.8224	0.0681	0.9292	19.647
	Wang <i>et al.</i>	0.0577	0.0495	33.816	0.9637	0.0439	0.9539	21.000
	Ours	<b>0.0150</b>	<b>0.0410</b>	<b>35.296</b>	<b>0.9841</b>	<b>0.0114</b>	<b>0.9887</b>	<b>4.000</b>

**Table 2**

Quantitative results for flow images in the converging channel geometry, comparing FFDNet [49], BM3D [23], MC-DCT [42], Wang *et al.* [37], and the proposed method.



**Figure 6.** Discrepancies in velocity fields among the noisy input (gray), reconstructed results (red for positive, blue for negative), and ground truth (green “+”). The upper indicates the velocity in x-axis while the lower for y-axis. (a) Low noise; (b) High noise.



**Figure 7.** Illustration of segmentation uncertainty for the converging channel geometry with Gaussian noise. The maps show the standard deviation computed from five noisy images with randomly generated noise. The left corresponds to the low-noise case, while the right represents the high-noise case.

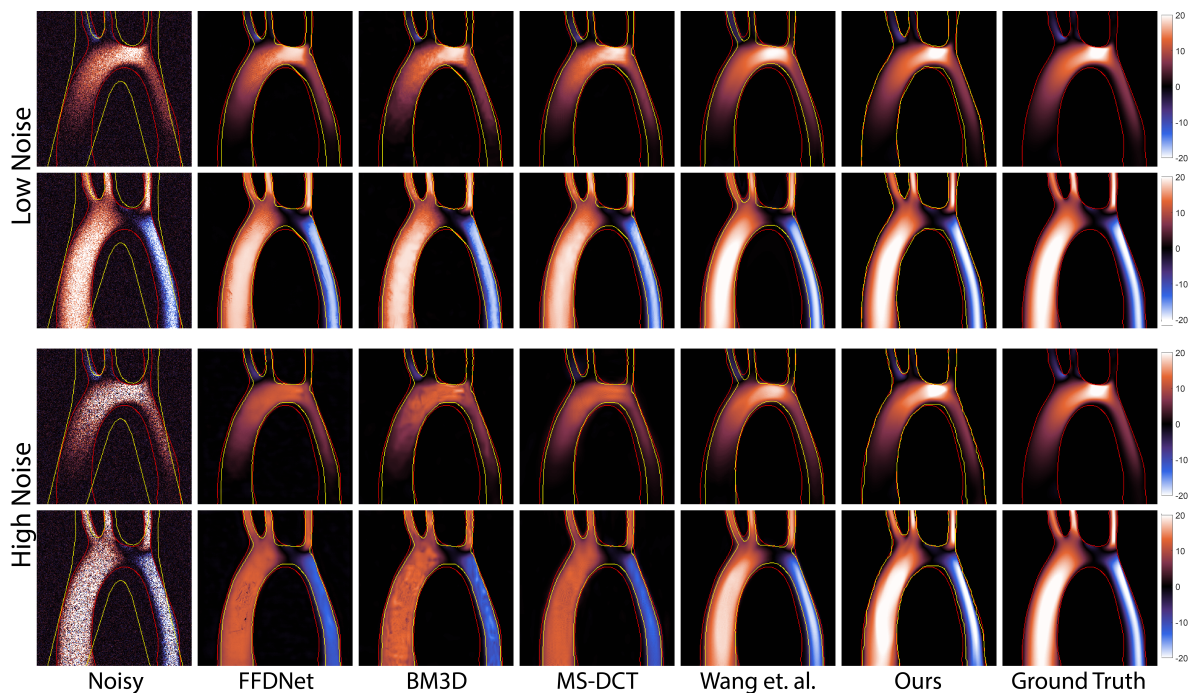
550 training process takes approximately 5 minutes for the converging channel geometry case.

551 We compared our method with learning-based approaches like FFDNet [49] and traditional  
 552 image processing techniques such as BM3D [23], multiscale DCT [42], and a physics-based  
 553 method [37]. The evaluation included both quantitative metrics and qualitative visualizations  
 554 for flow image reconstruction and region segmentation. Since competing methods do not  
 555 provide segmentations directly, we applied the geodesic active contour model [8] to extract  
 556 flow regions from their denoised outputs.

557 Figure 5 shows qualitative comparisons. The red contours indicate the ground-truth flow  
 558 regions, and the yellow contours show those inferred by each method. The first two rows  
 559 correspond to low-noise inputs, and the last two correspond to high-noise conditions. For  
 560 each noise level, the velocity components in the  $x$ - and  $y$ -directions are shown in the upper  
 561 and lower rows, respectively. The yellow contour in the "Noisy" column reflects the initial  
 562 domain estimate used to initialize the applicable methods.

563 As shown in Figure 5 and Table 2, our method consistently outperforms the others. Re-  
 564 garding flow image quality, conventional image denoising models are unable to fully eliminate  
 565 artifacts, often resulting in unnatural or physically implausible flow patterns. While the  
 566 physics-based approach by Wang *et al.* [37] achieves improved performance, its accuracy re-  
 567 mains limited in both mechanical reconstruction and domain segmentation, which arises from  
 568 the lack of coupled geometry evolution and the absence of domain boundary information dur-  
 569 ing the optimization process. In contrast, by incorporating the physics of fluid flow through the  
 570 Navier–Stokes equations and the associated domain evolution, our approach produces more  
 571 realistic and accurate reconstructions in both fluid mechanics and domain. The PINN model  
 572 with boundary information used for velocity estimation further ensures robust and efficient  
 573 optimization. Regarding region segmentation, it is observed that the active contour model,  
 574 when applied to denoised images, often yields inaccurate boundaries, particularly near wall  
 575 regions and close to the inlet, due to its reliance solely on image gradients. In contrast, our  
 576 method benefits from the underlying velocity field information through a registration-based  
 577 manner, enabling it to recover flow regions that closely match the ground truth and extract a  
 578 more accurate flow region.

579 To closely examine the noisy measurements, the reconstructed velocity field, and the  
 580 ground truth, we present Figure 6. The upper row shows the  $x$ -velocity component, and  
 581 the bottom row shows the  $y$ -velocity component. In each plot, the curves represent velocity



**Figure 8.** Qualitative comparison of flow and region reconstruction. Red contours show ground truth; yellow contours indicate predicted domains. Top two rows: low noise group; bottom two: high noise group. Each pair shows  $x$ - (top) and  $y$ -direction (bottom) velocity. Yellow contour in "Noisy" column marks the initialization.

582 magnitudes sampled along a horizontal line at a fixed  $x$ -location. The horizontal displacement from the vertical indicates speed — larger deviations correspond to higher velocities.  
 583 Reconstructed velocities are colored red for positive (rightward/upward) and blue for negative (leftward/downward) values; gray curves represent noisy observations, and green “+” markers indicate ground truth. From the profile observations, the reconstructed velocities closely  
 584 match the ground truth across inflow, outflow, and intermediate regions. They accurately capture the fluid dynamics and effectively suppress noise, as indicated by the gray curves,  
 585 while preserving physical consistency.  
 586

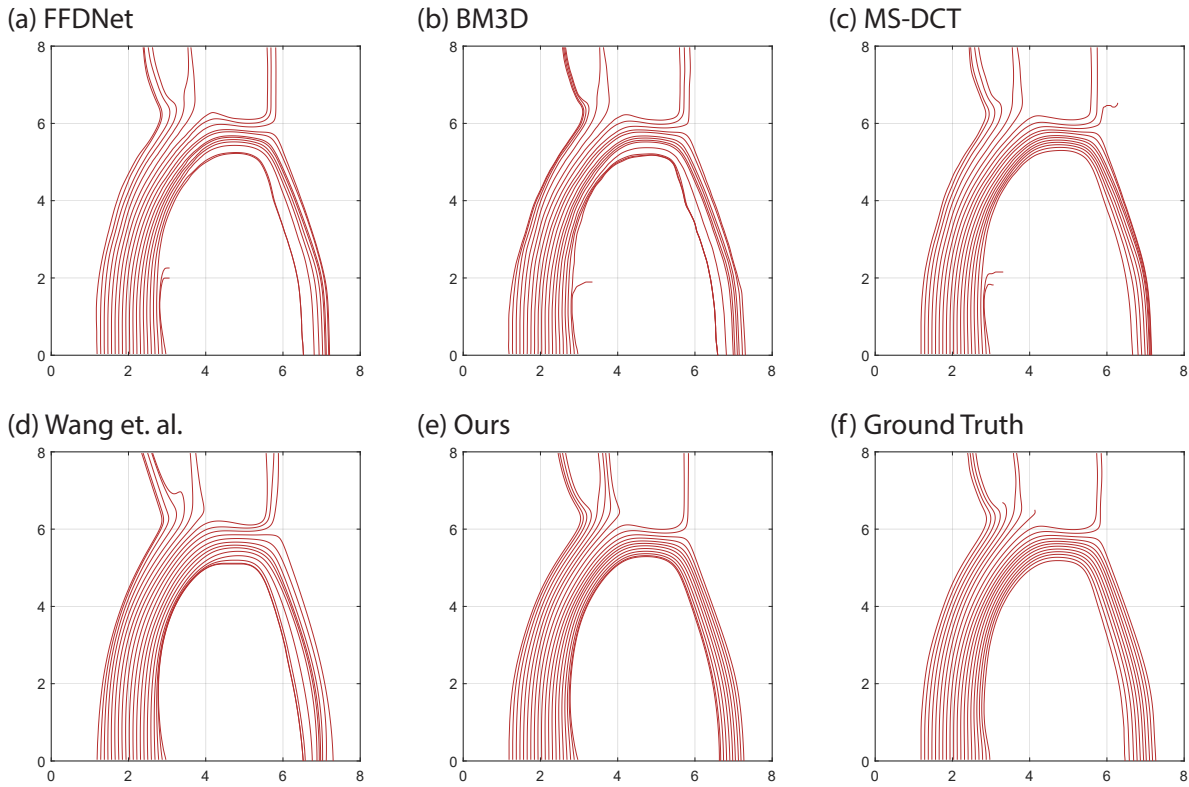
590 We also evaluate the uncertainty of our model under different levels of Gaussian noise. Specifically, we apply our method to five independently generated noisy images at both low  
 591 and high noise levels, and compute the standard deviation of the resulting segmented regions. The corresponding uncertainty map, shown in Figure 7, indicates that the segmentation un-  
 592 certainty remains relatively low even under high noise. This robustness can be attributed to  
 593 the global guidance provided by the registration-based geometry modeling.  
 594  
 595

596 **5.3. Flow Image over Aorta Vessel.** To evaluate our method in a more realistic setting,  
 597 we conducted experiments on synthetic flow images within an aorta geometry. The  $256 \times 256$   
 598 images represent an  $8 \text{ cm} \times 8 \text{ cm}$  domain, with the radius of the aorta around 2 cm. Following  
 599 the simulation setup in [36, 14], a parabolic inlet velocity profile with a peak of 20 cm/s

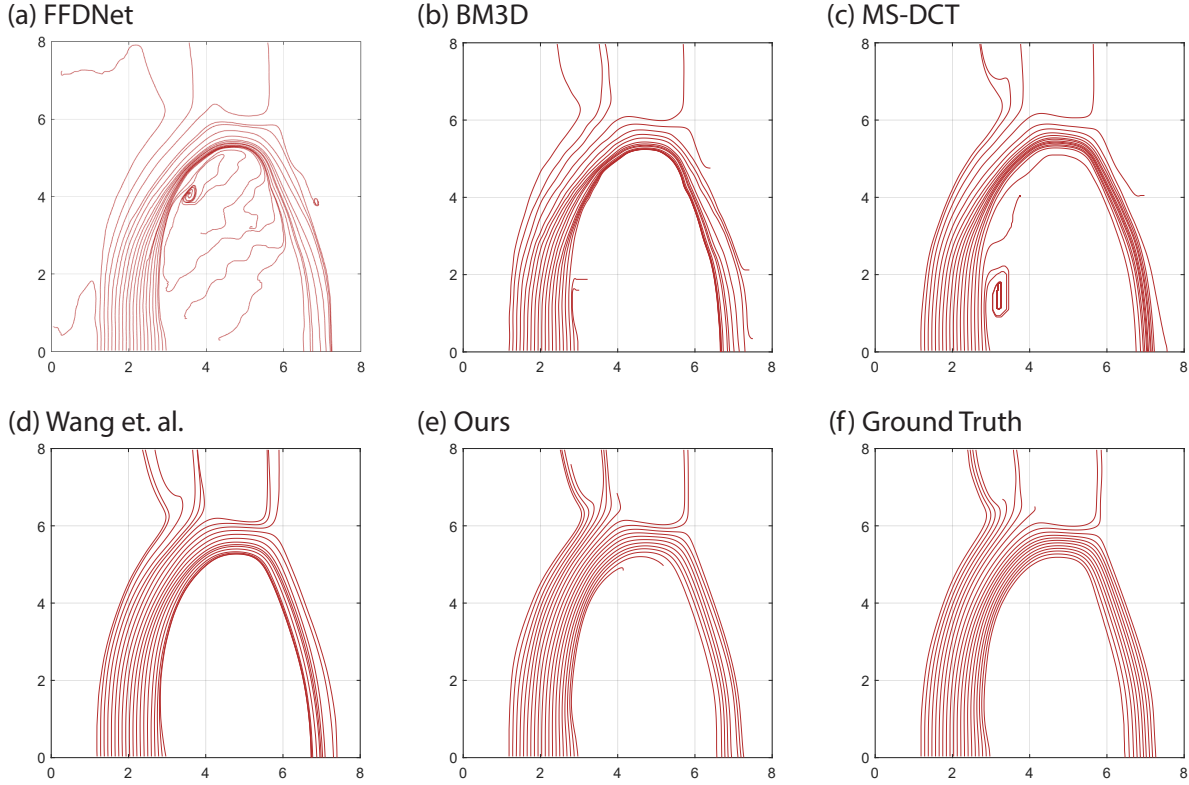
		Flow Image Reconstruction				Fluid Domain Segmentation		
		RE	MSE	PSNR	SSIM	MSE	Dice	HD95
Low	Noise	2.1716	5.8275	24.375	0.2967	0.1627	0.7904	26.476
	FFDNet	0.1456	0.8605	32.514	0.9251	0.0517	0.9194	15.454
	BM3D	0.1371	0.7917	32.276	0.9176	0.0529	0.9119	14.342
	MC-DCT	0.1670	0.7214	34.675	0.9353	0.0613	0.8922	14.000
	Wang <i>et al.</i>	0.1651	0.4825	35.642	0.9788	0.0587	0.9024	13.928
	Ours	<b>0.0847</b>	<b>0.4133</b>	<b>38.447</b>	<b>0.9869</b>	<b>0.0281</b>	<b>0.9574</b>	<b>8.000</b>
High	Noise	3.1875	24.0538	18.680	0.2487	0.1627	0.7904	26.476
	FFDNet	0.1566	3.4682	26.672	0.8915	0.0637	0.9019	16.170
	BM3D	0.2654	3.7285	24.287	0.8835	0.0607	0.8881	16.854
	MC-DCT	0.2753	4.1753	27.176	0.9117	0.0648	0.8767	15.000
	Wang <i>et al.</i>	0.2328	0.8734	33.374	0.9667	0.0761	0.8937	15.000
	Ours	<b>0.0964</b>	<b>0.5973</b>	<b>36.136</b>	<b>0.9804</b>	<b>0.0296</b>	<b>0.9540</b>	<b>8.000</b>

**Table 3**

Quantitative results for flow images in the aorta geometry, comparing FFDNet [49], BM3D [23], MC-DCT [42], Wang *et al.* [37], and the proposed method.



**Figure 9.** Streamline visualization reconstructed from a low-noise flow image under identical settings.

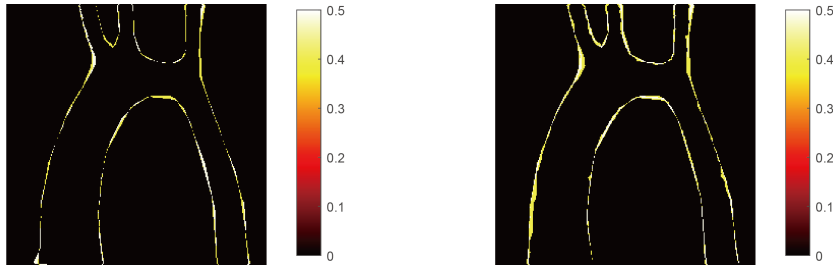


**Figure 10.** *Streamline visualization reconstructed from a high-noise flow image under identical settings.*

600 was used. The dynamic viscosity is set to  $\mu = 0.035 \text{ g} \cdot \text{cm}^{-1} \cdot \text{s}^{-1}$  and the density to  $\rho =$   
 601  $1.05 \text{ g} \cdot \text{cm}^{-3}$ . The Reynolds number for the simulation is 600, based on a characteristic length  
 602 of 1 cm and a characteristic velocity of 20 cm/s. Signal-dependent noise was added following  
 603 the model in (5.2). For low noise, the standard deviations were set to  $\sigma_1 = 1$  and  $\sigma_2 = 2$  for  
 604 the two directions, while for high noise, both directions used  $\sigma_1 = 2$  and  $\sigma_2 = 2$ . The training  
 605 was performed for 400 iterations, alternating between 40 iterations of the fluid subproblem  
 606 and 40 iterations of the geometry subproblem. Each fluid subproblem uses 50,000 collocation  
 607 points. The entire training process takes approximately 9 minutes for the aorta geometry  
 608 case.

609 As in the previous case, we compared our method with FFDNet [49], BM3D [23], mul-  
 610 tiscale DCT [42], and a physics-based method [37], evaluating both flow reconstruction and  
 611 segmentation. For methods without native segmentation, the geodesic active contour model  
 612 [8] was applied. Figure 8 presents qualitative results, following the same layout as Figure 5,  
 613 with red contours indicating ground truth and yellow contours showing inferred domains under  
 614 both low and high noise conditions.

615 As shown in the results in Figure 8 and Table 3, our method consistently outperforms the  
 616 others in all evaluation metrics. Conventional denoising methods such as FFDNet, BM3D, and  
 617 multiscale DCT struggle to effectively suppress noise, particularly under high-noise conditions.



**Figure 11.** Illustration of segmentation uncertainty for the aorta geometry with signal-dependent noise. The maps show the standard deviation computed from five noisy images with randomly generated noise. The left corresponds to the low-noise case, while the right represents the high-noise case.

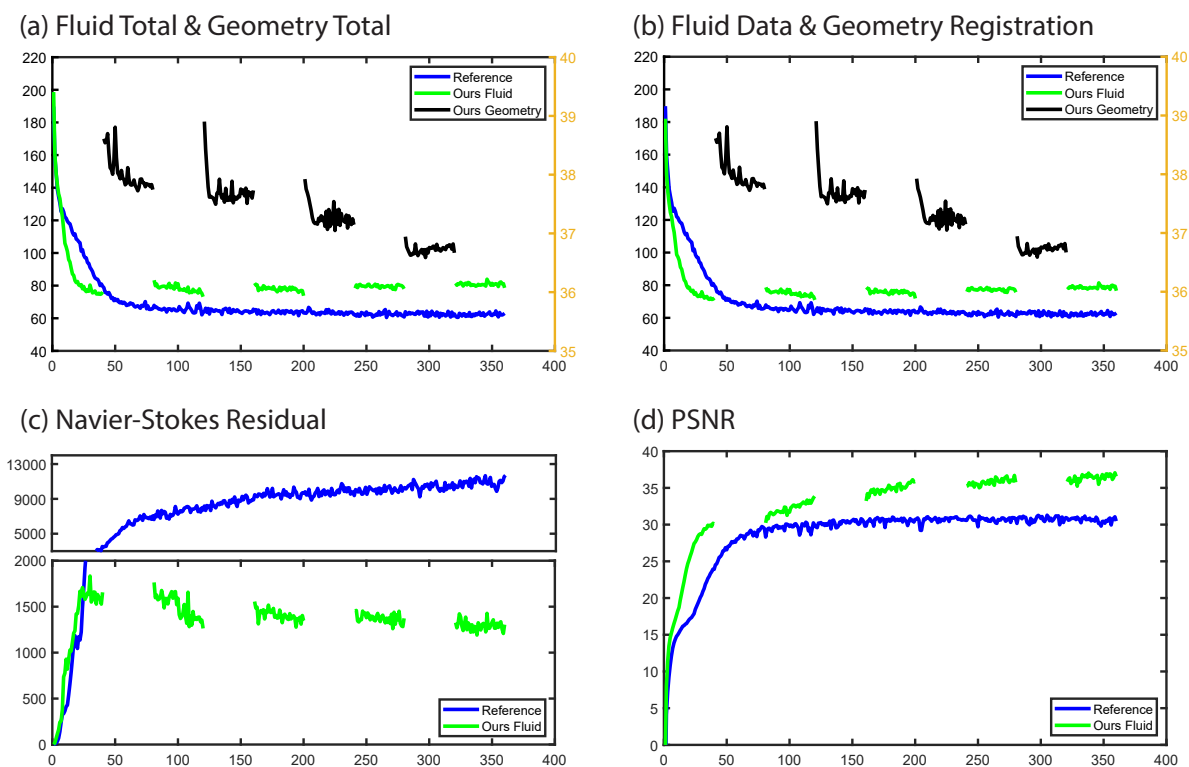
618 This limitation is especially evident in the last two columns of the figure, where these methods  
 619 produce flow fields with pronounced artifacts or physically implausible patterns. In such cases,  
 620 the reconstructed images fail to preserve the underlying flow structure and offer limited utility  
 621 for meaningful fluid dynamics analysis. While the physics-based approach by Wang *et al.* [37]  
 622 achieves improved quantitative performance compared to traditional denoising methods, its  
 623 accuracy remains insufficient, primarily due to the absence of coupled geometry evolution and  
 624 of the associated domain boundary information during the optimization process.

625 In contrast, our method consistently produces clean and physically realistic reconstruc-  
 626 tions, as demonstrated in Figures 9 and 10, which show reconstructed streamlines from low-  
 627 and high-noise images, respectively. This improvement is primarily attributed to the incor-  
 628 poration of physical constraints via the Navier–Stokes equations and the robust optimization  
 629 enabled by the differentiable PINN model, as well as the coupled domain evolution during the  
 630 process. By enforcing consistency with the governing physics of fluid flow and the domain,  
 631 our differentiable approach effectively suppresses noise while preserving essential flow features  
 632 embedded in noisy measurements.

633 Regarding flow region segmentation, intensity-based methods, such as geodesic active  
 634 contours, often result in inaccurate boundary detection, particularly near vessel walls, where  
 635 velocity naturally decreases. This misidentification leads to segmentations that deviate from  
 636 the true physiological flow region.

637 Our method, on the contrary, performs flow reconstruction and region segmentation in a  
 638 coupled manner, allowing each task to inform and refine the other. This mutual reinforcement  
 639 improves the accuracy of both the reconstructed flow fields and the segmented domains. The  
 640 synergy between physics-informed reconstruction and segmentation contributes to the overall  
 641 performance advantage of our approach, as confirmed by both visual results and quantitative  
 642 evaluations. For the high-noise aorta case, the segmented domain occasionally shows jagged  
 643 boundaries. This was addressed in [21] by adding a viscous term that implies motion of the  
 644 domain’s boundary by its mean curvature [28]. Instead, we apply morphological opening  
 645 followed by closing with a disk of radius  $r = 6$ , which effectively bounds the absolute value of  
 646 the boundary curvature by  $1/r$ .

647 To assess the segmentation uncertainty in the aorta geometry case with signal-dependent  
 648 noise, we follow a procedure similar to the converging channel experiment. Our method is  
 649 applied to five independently generated noisy images at both low and high noise levels, and the



**Figure 12.** Training dynamics and convergence behavior for the aorta geometry case by our method and the reference. The reference method is done by removing the Navier–Stokes residual terms and boundary conditions, training the model solely with the data-fidelity term. The green curve represents the total loss of the fluid subproblem, the black curve corresponds to the geometry subproblem, and the blue curve denotes the result obtained by the reference model trained with only the data-fidelity term. (a) Total fluid and geometry losses; (b) data loss in the fluid subproblem and registration loss in the geometry subproblem; (c) Navier–Stokes residuals; (d) PSNR evolution during training.

650 standard deviation of the resulting segmented regions is calculated. The uncertainty map in  
 651 Figure 11 shows that the segmentation remains stable, with controllable variations even under  
 652 high noise conditions. This consistency highlights the robustness of the proposed framework,  
 653 benefiting from the globally guided deformation in the registration-based geometry modeling.

654 We further examine the error plots in Figure 12 to better understand the training dynamics  
 655 and the role of the physics-informed component, by comparing our method with a reference  
 656 approach. The reference method is obtained by removing the Navier–Stokes residual terms  
 657 and boundary conditions, and training the model solely using the data-fidelity term. In these  
 658 plots, for our method, the green curve represents the total loss of the fluid subproblem, while  
 659 the black curve corresponds to the geometry subproblem. The blue curve shows the result  
 660 obtained using the reference approach. As illustrated in Figure 12(a), the alternating behavior  
 661 between the total fluid loss (y-axis on the left) and the total geometry loss (y-axis on the right)  
 662 clearly demonstrates the effect of our alternating optimization strategy. Because our method

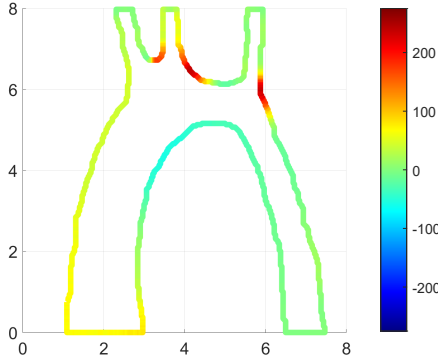


Figure 13. Wall shear stress (WSS) distribution on the aortic wall obtained from the reconstructed solution in the high-noise case. The color codes the spatial variation of shear stress magnitude ( $\text{dyn}/\text{cm}^2$ ).

663 incorporates physical constraints, the data error decreases more slowly than in the reference  
 664 model. However, as shown in Figure 12(c), the minimization of the Navier–Stokes residual  
 665 leads to significantly improved reconstruction accuracy, which is reflected by the better PSNR  
 666 in Figure 12(d).

667 Wall shear stress (WSS) is a clinically significant hemodynamic quantity. Therefore, we  
 668 further estimate this variable and visualize its spatial distribution in Figure 13. The results are  
 669 obtained from the high-noise aorta geometry case. Using the reconstructed velocity networks,  
 670 which allow direct computation of velocity gradients, together with the pressure network, we  
 671 evaluate the WSS distribution on the vessel boundary. A color-coded plot with an accom-  
 672 panying color bar is provided to illustrate its spatial variation. Combined with the velocity  
 673 visualization shown in Figure 8, we observe that elevated shear stress occurs in regions where  
 674 the flow direction changes markedly due to geometric constraints of the vessel, while lower  
 675 WSS values appear in areas where the velocity field propagates more smoothly.

676 **5.4. Self Ablation.** Here, we perform ablation studies to assess the impact of different  
 677 weighting parameters in the optimization of the fluid and geometry subproblems. The high-  
 678 noise flow image with Aorta geometry is used.

679 **Test on super-resolution** Our method is built upon a meshless PINN solver constrained  
 680 by the Navier–Stokes equations. Thanks to the meshless formulation, the reconstructed veloc-  
 681 ity field can be evaluated at arbitrary spatial resolution while maintaining solution accuracy.  
 682 To demonstrate this advantage, we include a super-resolution experiment using aorta flow  
 683 data of sizes  $64 \times 64$ ,  $128 \times 128$ , and the originally  $256 \times 256$ . For each case, we apply the same  
 684 reconstruction algorithm, and then evaluate the resulting velocity field on a  $256 \times 256$  grid. As  
 685 shown in Table 4, all reconstructed high-resolution velocity fields closely match the  $256 \times 256$   
 686 ground truth, as reflected by the quantitative metrics. This experiment confirms that our  
 687 method can reliably recover high-quality flow fields even from low-resolution measurements.

688 **Test on  $\alpha_{\text{data}}$**  : We conducted an ablation study to evaluate the effect of the data fitting  
 689 weight parameter  $\alpha_{\text{data}}$ , which plays a critical role in aligning the predicted velocity field with  
 690 the observed data. The best overall performance was achieved at  $\alpha_{\text{data}} = 10^0$ , striking a  
 691 strong yet balanced emphasis on data fidelity while maintaining physical regularization. At

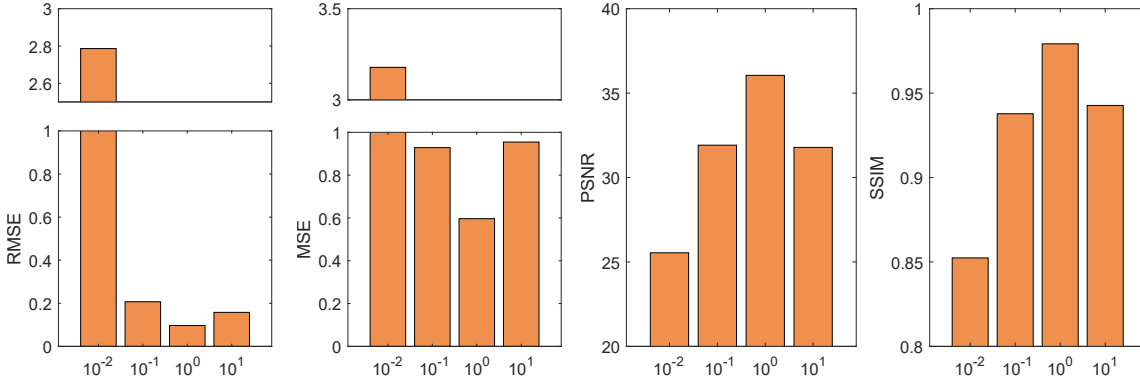


Figure 14. Ablation study evaluating the impact of the data fitting weight  $\alpha_{data}$  on training performance.

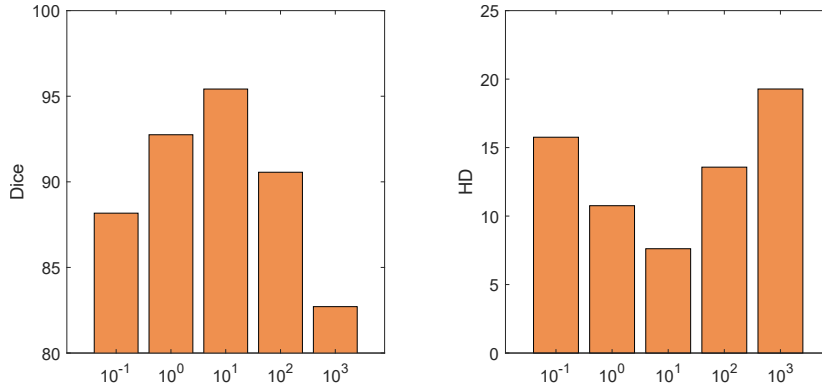


Figure 15. Ablation study evaluating the impact of Beltrami regularization weight  $\alpha_{data}$  on training performance.

692  $\alpha_{data} = 10^1$ , the model still performed well, although signs of overfitting appeared as the  
693 network prioritized fitting noisy data, which reduced generalization. Lowering the weight to  
694  $10^{-1}$  caused a slight performance drop due to weaker data alignment, particularly in regions  
695 with fine flow features. When  $\alpha_{data}$  was reduced further to  $10^{-2}$ , the performance suffered  
696 significantly: the data constraint was too weak to effectively guide the model, causing the  
697 optimization to rely primarily on regularization and resulting in poor recovery of the flow  
698 field and domain structure.

699 **Test on  $\alpha_{bc}$  :** We conducted a self-ablation study on the sensitivity to the Beltrami  
700 coefficient regularization weight,  $\alpha_{bc}$ . The best performance occurred at  $\alpha_{bc} = 10^1$ , balancing  
701 mapping regularization and registration. At  $\alpha_{bc} = 10^0$ , results were acceptable, although mi-  
702 nor deterioration appeared, probably due to increased deformation causing localized shrinkage.  
703 Lowering to  $\alpha_{bc} = 10^{-1}$  significantly degraded performance, as weak regularization allowed  
704 excessive domain deviation, topological errors, and outlier regions that harmed precision. Con-  
705 versely, at  $\alpha_{bc} = 10^2$ , over-regularization constrained the model, limiting domain propagation  
706 and adaptation to subtle variations, leading to incomplete or inaccurate reconstructions. This  
707 over-constraint effect intensified at  $\alpha_{bc} = 10^3$ .

Resol.	RE	MSE	PSNR	SSIM
$64^2$	0.1381	0.6891	33.648	0.9498
$128^2$	0.1207	0.6343	35.187	0.9679
$256^2$	0.0969	0.5966	36.053	0.9792
$\alpha_{\text{data}}$	RE	MSE	PSNR	SSIM
$10^{-2}$	2.7861	3.1784	25.542	0.8524
$10^{-1}$	0.2071	0.9284	31.917	0.9378
$10^0$	0.0969	0.5966	36.053	0.9792
$10^1$	0.1578	0.9546	31.786	0.9427
$\alpha_{\text{bc}}$	$\ \mu\ _2$	Dice	HD	$\ \mu\ _\infty < 1$
$10^{-1}$	1.078	0.8817	15.757	✗
$10^0$	0.337	0.9375	10.757	✓
$10^1$	0.068	0.9542	7.616	✓
$10^2$	0.005	0.9056	13.572	✓
$10^3$	0.000	0.8271	19.278	✓

**Table 4**

*Ablation study on the impact of flow image resolution, varying data fidelity weight  $\alpha_{\text{data}}$  and the Beltrami regularization weight  $\alpha_{\text{bc}}$  for flow images in an aorta-shaped domain under high-noise conditions.*

708 **6. Conclusion.** In this work, we propose a novel framework that integrates deep learning  
709 with physical modeling to improve the reconstruction of 2D flow images under noisy measure-  
710 ments. The problem is formulated to enforce physical consistency through the Navier–Stokes  
711 equations and preserve anatomical structures using quasi-conformal mappings. It is decom-  
712 posed into two subproblems: a fluid subproblem, which estimates the velocity field within a  
713 given domain using Navier–Stokes constraints, and a geometry subproblem, which refines the  
714 domain by aligning the reconstructed velocity with observed data via a quasi-conformal map-  
715 ping. These subproblems are solved iteratively in a Gauss–Seidel fashion until convergence,  
716 providing a solution supported by theoretical guarantees. In practice, we employ a Physics-  
717 Informed Neural Network for the fluid subproblem and a UNet-based Mapping Estimator for  
718 the geometry subproblem.

719 We extensively validated our method through experiments. Initial tests on synthetic  
720 flow images in a converging channel under varying Gaussian noise levels validate the proposed  
721 method. Further evaluations on anatomically realistic aorta geometries with signal-dependent  
722 noise confirmed its effectiveness in complex real-world settings. Our method consistently out-  
723 performed the baseline approaches, delivering more accurate and higher-quality reconstruc-  
724 tions. Ablation studies also highlighted the impact of weighting schemes and guided the  
725 selection of effective hyperparameters.

726 Future work will extend this framework to 3D flow reconstruction and 4D dynamic flow  
727 analysis by replacing the 2D quasi-conformal mappings with Jacobian-based constraints or  
728 hyperelastic energy models to ensure smooth and bijective deformations in higher dimensions.  
729 **We also plan to adapt the framework for higher Reynolds number flows by employing more**  
730 **expressive architectures, such as PirateNets [38].** Additionally, we aim to tackle more chal-

731 lenging imaging modalities, such as ultrasound Doppler imaging, which involve higher noise  
732 levels and real-time computational constraints. Furthermore, since the current Gauss–Seidel  
733 alternating scheme lacks a formal convergence guarantee, and the validation of the split prob-  
734 lems relies primarily on empirical observations, we plan to develop a theoretical convergence  
735 analysis by adding additional regularization.

736

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850 **Appendix A. Proof For Theorem 3.3.**

851 *Proof.*

852 **Conclusion (a):**  $c_n \rightarrow \text{id}$  in  $L^\infty(D)$ .

853 Since each  $f_n$  is a diffeomorphism of  $D$ , the limit  $f_*$  is also a diffeomorphism. Thus,  $f_n \rightarrow f_*$   
854 uniformly on  $D$ .

855 By Lemma A.1, we have  $c_n \rightarrow \text{id}$  in  $L^\infty(D)$ . This implies that the objective functional  
856  $\|\tilde{u} - u_n \circ c_n\|_{L^2(D)}$  converges to  $\|\tilde{u} - u\|_{L^2(D)}^2$  as the iterations proceed given that  $D$  is compact.  
857 Such convergence reflects the diminishing need for geometric correction.

858 **Conclusion (b) -  $f_*$  is a partial and local minimizer**

859 From assumption (1) and (2), and  $u_n = \mathcal{U}[f_n]$  by definition, we conclude that

$$860 \quad \mathcal{U}[f_n] \rightarrow \mathcal{U}[f_*] = u_* \quad \text{in } L^\infty(D),$$

861 which implies

$$862 \quad J_{\mathcal{U}}(f_n) = \|\tilde{u} - \mathcal{U}[f_n]\|_{L^2(D)} \rightarrow \|\tilde{u} - \mathcal{U}[f_*]\|_{L^2(D)} = J_{\mathcal{U}}(f_*).$$

863 **- Step 1:**

864 Let  $\varepsilon > 0$  be arbitrary. By assumption (1),  $u_n \rightarrow u_*$  in  $L^\infty(D)$ . So there exists  $N_1$  such that  
865 for all  $n \geq N_1$ ,

$$866 \quad \|u_n - u_*\|_{L^\infty(D)} < \frac{\varepsilon}{3}.$$

867 Now fix a small perturbation  $c \in \mathcal{H}(D)$ , close to identity, which we will quantify later. Define

$$868 \quad f = c^{-1} \circ f_n \quad \text{and} \quad \mathcal{U}[f] := \mathcal{U}[c^{-1} \circ f_n].$$

869 With  $c_n = \arg \min_{c \in \mathcal{H}(D)} \|\tilde{u} - u_n \circ c\|_{L^2(D)}$ , we have

$$870 \quad (\text{A.1}) \quad \|\tilde{u} - u_n \circ c_n\|_{L^2(D)} \leq \|\tilde{u} - u_n \circ c\|_{L^2(D)}, \quad \forall c \in \mathcal{H}(D).$$

871 **- Step 2: Inequality for  $\|\mathcal{U}[c^{-1} \circ f_n] - \mathcal{U}[f_n]\|_{L^2(D)}$**

872 By the continuity of the mapping in assumption (3), for every  $\varepsilon > 0$ , there exists  $\delta > 0$  such  
873 that for  $f, f_n \in \mathcal{H}(D)$ ,

$$874 \quad \|f - f_n\|_{L^\infty(D)} < \delta \quad \Rightarrow \quad \|\mathcal{U}[f] - \mathcal{U}[f_n]\|_{L^2(D)} < \frac{\varepsilon}{3}.$$

875 By conclusion (a), we have  $\|c^{-1} \circ f_n - f_n\|_{L^\infty(D)} < \delta$  for  $n$  large enough. Hence, we get

$$876 \quad (\text{A.2}) \quad \|\mathcal{U}[c^{-1} \circ f_n] - \mathcal{U}[f_n]\|_{L^2(D)} < \frac{\varepsilon}{3}.$$

877 **- Step 3: Inequality for  $\|u_n \circ c - u_n\|_{L^2(D)}$**

878 As  $u_n$  satisfies (3.7) for fixed  $f_n$ , with assumption (4), there is a constant  $M$  such that

$$879 \quad |u_n(c(x)) - u_n(x)|^2 \leq M^2 |c(x) - x|^2.$$

880 Integrating over  $D$ , we get

$$881 \quad \|u_n \circ c - u_n\|_{L^2(D)}^2 = \int_D |u_n(c(x)) - u_n(x)|^2 dx \leq M^2 \int_D |c(x) - x|^2 dx$$

882 Now since  $\|c - \text{id}\|_{L^\infty(D)} < \delta$ , if we choose

$$883 \quad \delta < \frac{\varepsilon}{3M\sqrt{|D|}}$$

884 We can have

$$885 \quad (\text{A.3}) \quad \|u_n \circ c - u_n\|_{L^2(D)} < M\delta\sqrt{|D|} < \frac{\varepsilon}{3}$$

886 **- Step 4:**

887 With (A.2) and (A.3), we have

$$888 \quad (\text{A.4}) \quad \|u_n \circ c - \mathcal{U}[c^{-1} \circ f_n]\|_{L^2(D)} \leq \|u_n \circ c - u_n\|_{L^2(D)} + \|u_n - \mathcal{U}[c^{-1} \circ f_n]\|_{L^2(D)} < \frac{2\varepsilon}{3}.$$

889 We now compare

$$890 \quad J_{\mathcal{U}}(f) = \|\tilde{u} - \mathcal{U}[c^{-1} \circ f_n]\|_{L^2(D)}, \quad J_{\mathcal{U}}(f_{n+1}) = \|\tilde{u} - \mathcal{U}[c_n^{-1} \circ f_n]\|_{L^2(D)}.$$

891 By the triangle inequality

$$892 \quad \|\tilde{u} - \mathcal{U}[c^{-1} \circ f_n]\| \geq \|\tilde{u} - u_n \circ c\| - \|u_n \circ c - \mathcal{U}[c^{-1} \circ f_n]\|.$$

893 Using (A.4)

$$894 \quad \|\tilde{u} - \mathcal{U}[c^{-1} \circ f_n]\| > \|\tilde{u} - u_n \circ c\| - \frac{2\varepsilon}{3}.$$

895 Together with (A.1), we have

$$896 \quad (\text{A.5}) \quad J_{\mathcal{U}}(f) \geq \|\tilde{u} - u_n \circ c_n\| - \frac{2\varepsilon}{3}.$$

897 Similarly, we can have  $\|u_n \circ c_n - \mathcal{U}[c_n^{-1} \circ f_n]\| < \varepsilon/3$  by the techniques as above, and obtain

$$898 \quad (\text{A.6}) \quad \|\tilde{u} - u_n \circ c_n\| \geq \|\tilde{u} - \mathcal{U}[f_{n+1}]\| - \|\mathcal{U}[f_{n+1}] - u_n \circ c_n\| > J(f_{n+1}) - \frac{\varepsilon}{3}.$$

899 Hence

$$900 \quad J_{\mathcal{U}}(f) > J_{\mathcal{U}}(f_{n+1}) - \varepsilon.$$

901 Take  $n \rightarrow \infty$ , then  $f_{n+1} \rightarrow f_*$ ,  $\mathcal{U}[f_{n+1}] \rightarrow \mathcal{U}[f_*]$ , so

$$902 \quad (\text{A.7}) \quad \liminf_{f \rightarrow f_*} J_{\mathcal{U}}(f) \geq J_{\mathcal{U}}(f_*).$$

903 Thus,  $f_*$  is a partial and local minimizer of the misfit.

904 **Conclusion (c) -  $u_*$  is a partial and local minimizer**

905 Denote the solution set for  $u$  as

$$906 \quad \mathcal{A} := \{\mathcal{U}[f] \in L^2(D) \mid f \in \mathcal{H}(D)\}.$$

907 Define the functional as

$$908 \quad J(u) = \|\tilde{u} - u\|_{L^2(D)}.$$

909 As  $f_* \in \mathcal{H}(D)$  is a local minimizer for  $f \mapsto J(\mathcal{U}[f])$ , there exists  $\delta_1 > 0$  such that for all  
910  $f \in \mathcal{H}(D)$  and  $\|f - f_*\|_{L^\infty(D)} < \delta_1$  such that

$$911 \quad (\text{A.8}) \quad J(\mathcal{U}[f]) = \|\tilde{u} - \mathcal{U}[f]\|_{L^2(D)} \geq \|\tilde{u} - \mathcal{U}[f_*]\|_{L^2(D)} = J(u_*).$$

Let  $\varepsilon > 0$ , by assumption (3), there exists  $\delta_2 > 0$  such that for all  $f \in \mathcal{H}(D)$ , which satisfy  
 $\|f - f_*\|_{L^\infty(D)} < \delta_2$ , we have

$$\|\mathcal{U}[f] - \mathcal{U}[f_*]\|_{L^2(D)} < \varepsilon,$$

912 where we can make  $\delta_2 < \delta_1$  without loss of generality.

913 Then, for any  $u \in \mathcal{A}$  and  $\|\mathcal{U}[f] - \mathcal{U}[f_*]\|_{L^2(D)} < \varepsilon$ , we have

$$914 \quad \|f - f_*\|_{L^\infty(D)} < \delta_2 < \delta_1.$$

915 By (A.8), we have

$$916 \quad (\text{A.9}) \quad J(u) = J(\mathcal{U}[f]) \geq J(\mathcal{U}[f_*]) = J(u_*),$$

917 which means  $u_*$  is a partial and local minimizer of  $J$  over  $\mathcal{A}$ . ■

918 In the theory below, we show that if the sequence of mappings  $f_n$ , obtained from some  
919 iterative algorithm for minimizing  $\|\tilde{u} - u\|_{L^2(D)}^2$  under (3.7), converges, then the corresponding  
920 correction mappings  $c_n = f_{n+1} \circ f_n^{-1}$  will converge to the identity mapping.

921 **Lemma A.1.** *Let  $D \subset \mathbb{C}$  be a compact set. Given a sequence of diffeomorphisms  $f_n : D \rightarrow$   
922  $D$ . We have*

$$923 \quad (\text{A.10}) \quad f_n \rightarrow f \quad \text{uniformly on } D, \text{ where } f : D \rightarrow D \text{ is a diffeomorphism,}$$

924 *if and only if*

$$925 \quad (\text{A.11}) \quad f_{n+1} \circ f_n^{-1} \rightarrow \text{id}_D \quad \text{uniformly on } D.$$

926

927 *Proof.*

928 ( $\Rightarrow$ ):

929 Let  $\varepsilon > 0$  be arbitrary. Since  $f_n \rightarrow f$  uniformly on  $D$ , there exists  $N_1 \in \mathbb{N}$  such that for all  
 930  $n \geq N_1$ ,

$$931 \quad (\text{A.12}) \quad \sup_{x \in D} \|f_n(x) - f(x)\| < \frac{\varepsilon}{2}.$$

932 Additionally, the uniform convergence of  $f_n \rightarrow f$  and the fact that each  $f_n$  and  $f$  are diffeo-  
 933 morphisms implies that  $f_n^{-1} \rightarrow f^{-1}$  uniformly on  $D$  by Lemma A.2.

934 Let  $N = \max(N_1, N_2)$ . For all  $n \geq N$  and all  $x \in D$ , define  $y := f_n^{-1}(x)$ . Then  $x = f_n(y)$ ,  
 935 and we compute

$$936 \quad (\text{A.13}) \quad f_{n+1} \circ f_n^{-1}(x) = f_{n+1}(y),$$

937 so that

$$938 \quad (\text{A.14}) \quad \|f_{n+1} \circ f_n^{-1}(x) - x\| = \|f_{n+1}(y) - f_n(y)\|.$$

939 Now, we insert  $f(y)$  and use the triangle inequality

$$940 \quad (\text{A.15}) \quad \|f_{n+1}(y) - f_n(y)\| \leq \|f_{n+1}(y) - f(y)\| + \|f(y) - f_n(y)\| < \frac{\varepsilon}{2} + \frac{\varepsilon}{2} = \varepsilon.$$

941 Since this holds for all  $x \in D$ , we have

$$942 \quad (\text{A.16}) \quad \sup_{x \in D} \|f_{n+1} \circ f_n^{-1}(x) - x\| < \varepsilon, \quad \forall n \geq N.$$

943 Hence,

$$944 \quad (\text{A.17}) \quad f_{n+1} \circ f_n^{-1} \rightarrow \mathbf{id}_D \quad \text{uniformly on } D.$$

945 ( $\Leftarrow$ ):

946 With  $f_{n+1} \circ f_n^{-1} \rightarrow \mathbf{id}_D$  uniformly on  $D$  for every  $\varepsilon > 0$ , there exists  $N \in \mathbb{N}$  such that for all  
 947  $n \geq N$  and all  $x \in D$ ,

$$948 \quad (\text{A.18}) \quad |f_{n+1}(f_n^{-1}(x)) - x| < \varepsilon.$$

949 Let  $y = f_n^{-1}(x)$ , so  $x = f_n(y)$ . Then the above becomes

$$950 \quad (\text{A.19}) \quad |f_{n+1}(y) - f_n(y)| < \varepsilon, \quad \forall y \in D.$$

951 Thus,

$$952 \quad (\text{A.20}) \quad \sup_{y \in D} |f_{n+1}(y) - f_n(y)| < \varepsilon.$$

953 This shows that the sequence  $\{f_n\}$  is uniformly Cauchy in  $C(D)$ , the space of continuous  
 954 functions on the compact set  $D$  with the sup norm.

955 Since  $D$  is compact and each  $f_n$  is continuous, there exists a continuous function  $f : D \rightarrow D$   
 956 such that

$$957 \quad (\text{A.21}) \quad f_n \rightarrow f \quad \text{uniformly on } D. \quad \blacksquare$$

958 **Lemma A.2.** Let  $D \subset \mathbb{C}$  be a compact set, and let  $\{f_n\}_{n=1}^{\infty}$  be a sequence of diffeomorphism  
 959 from  $D$  onto itself such that  $f_n \rightarrow f$  uniformly on  $D$ . Assume that the limit function  $f : D \rightarrow$   
 960  $D$  is also a diffeomorphism. Then

$$961 \quad (\text{A.22}) \quad f_n^{-1} \rightarrow f^{-1} \quad \text{uniformly on } D.$$

962

963 *Proof.* For any  $\varepsilon > 0$ . Since  $f_n^{-1}$  is continuous on the compact set  $D$ , there exists  $\delta > 0$   
 964 such that for any  $x_1, x_2 \in D$ ,

$$965 \quad (\text{A.23}) \quad |x_1 - x_2| < \delta \quad \Rightarrow \quad |f_n^{-1}(x_1) - f_n^{-1}(x_2)| < \varepsilon.$$

966 Now fix any  $x \in D$ , and let  $y = f^{-1}(x)$ . We can have  $x = f(y)$  and  $x_n := f_n(y)$ . For  $\delta$   
 967 mentioned above, we can find a sufficiently large  $n$ , as  $f_n \rightarrow f$  uniformly, to obtain

$$968 \quad (\text{A.24}) \quad |x_n - x| = |f_n(y) - f(y)| < \delta.$$

969 Since  $f_n$  is bijective, we have  $f_n^{-1}(x_n) = y = f^{-1}(x)$ . Together with (A.23), we have

$$970 \quad (\text{A.25}) \quad |f_n^{-1}(x) - f^{-1}(x)| \leq |f_n^{-1}(x) - f_n^{-1}(x_n)| + |f_n^{-1}(x_n) - f^{-1}(x)| \\ < \varepsilon + 0 = \varepsilon.$$

971 Therefore,  $f_n^{-1} \rightarrow f^{-1}$  uniformly on  $D$ . ■

## 972 **Appendix B. Metrics Definition.**

973 To evaluate the accuracy and speed of our methods, some accuracy measurements are  
 974 reported in the following sections.

975 The relative error (RE) is defined as

$$976 \quad (\text{B.1}) \quad \text{RE}(\mathbf{u}^*, \mathbf{u}^{\text{GT}}) = \sum_{\mathbf{x} \in \mathcal{G}} \frac{|\mathbf{u}^*(\mathbf{x}) - \mathbf{u}^{\text{GT}}(\mathbf{x})|^2}{|\mathbf{u}^{\text{GT}}(\mathbf{x})|^2},$$

977 where  $\mathbf{u}^*$  is the reconstructed flow,  $\mathbf{u}^{\text{GT}}$  is the ground truth flow, and  $\mathcal{G}$  denotes the set of  
 978 pixel points.

979 Peak Signal-to-Noise Ratio (PSNR) is defined as

$$980 \quad (\text{B.2}) \quad \text{PSNR}(\mathbf{u}^*, \mathbf{u}^{\text{GT}}) = 10 \log_{10} \left( \frac{\max_{\mathbf{x} \in \mathcal{G}} (\mathbf{u}^{\text{GT}}(\mathbf{x}))^2}{\frac{1}{|\mathcal{G}|} \sum_{\mathbf{x} \in \mathcal{G}} |\mathbf{u}^*(\mathbf{x}) - \mathbf{u}^{\text{GT}}(\mathbf{x})|^2} \right),$$

981 where  $|\mathcal{G}|$  is number of points in  $\mathcal{G}$ .

982 Structural Similarity Index (SSIM) is defined as

$$983 \quad (\text{B.3}) \quad \text{SSIM}(\mathbf{u}^*, \mathbf{u}^{\text{GT}}) = \frac{1}{|\mathcal{G}|} \sum_{\mathbf{x} \in \mathcal{G}} \frac{(2\mu_{\mathbf{u}^*}(\mathbf{x})\mu_{\mathbf{u}^{\text{GT}}}(\mathbf{x}) + C_1) (2\sigma_{\mathbf{u}^*, \mathbf{u}^{\text{GT}}}(\mathbf{x}) + C_2)}{(\mu_{\mathbf{u}^*}(\mathbf{x})^2 + \mu_{\mathbf{u}^{\text{GT}}}(\mathbf{x})^2 + C_1) (\sigma_{\mathbf{u}^*}(\mathbf{x})^2 + \sigma_{\mathbf{u}^{\text{GT}}}(\mathbf{x})^2 + C_2)},$$

984 where  $C_1, C_2$  are small constants to stabilize the division, and

$$\begin{aligned}
985 \quad \mu_{\mathbf{u}^*} &= \mathbf{u}^* * w, & \sigma_{\mathbf{u}^*}^2 &= (\mathbf{u}^*)^2 * w - (\mu_{\mathbf{u}^*})^2, \\
986 \quad \mu_{\mathbf{u}^{\text{GT}}} &= \mathbf{u}^{\text{GT}} * w, & \sigma_{\mathbf{u}^{\text{GT}}}^2 &= (\mathbf{u}^{\text{GT}})^2 * w - (\mu_{\mathbf{u}^{\text{GT}}})^2, \\
987 \quad \sigma_{\mathbf{u}^*, \mathbf{u}^{\text{GT}}} &= (\mathbf{u}^* \cdot \mathbf{u}^{\text{GT}}) * w - \mu_{\mathbf{u}^*} \cdot \mu_{\mathbf{u}^{\text{GT}}},
\end{aligned}$$

988 where  $\omega$  is a Gaussian kernel of mean as 0 and standard deviation as 1.5. The convolution is  
989 with reflect padding

990 The Dice coefficient measures the spatial overlap of two binary masks and is defined as

$$991 \quad (\text{B.4}) \quad \text{Dice}(\Omega_*, \Omega_{\text{GT}}) = \frac{2|\Omega_* \cap \Omega_{\text{GT}}|}{|\Omega_*| + |\Omega_{\text{GT}}|},$$

992 where  $\Omega_*$  and  $\Omega_{\text{GT}}$  represent the predicted and ground-truth regions.  $|\Omega|$  is the area of  $\Omega$ . A  
993 value of 1 indicates perfect overlap.

994 The 95th percentile Hausdorff distance (HD95) measures the boundary discrepancy be-  
995 tween two segmented regions by taking the 95th percentile of the bidirectional surface dis-  
996 tances, thereby reducing the influence of outlier points or noisy boundaries. It is defined as  
997

$$998 \quad (\text{B.5}) \quad \text{HD}_{95}(\Omega_*, \Omega_{\text{GT}}) = \max \left\{ \text{P}_{95}(\{d(x, \Omega_{\text{GT}}) \mid x \in \Omega_*\}), \text{P}_{95}(\{d(y, \Omega_*) \mid y \in \Omega_{\text{GT}}\}) \right\},$$

999 where  $d(x, \Omega_{\text{GT}}) = \inf_{y \in \Omega_{\text{GT}}} \|x - y\|$ ,  $d(y, \Omega_*) = \inf_{x \in \Omega_*} \|y - x\|$  and  $\text{P}_{95}(\{z_i\})$  denotes the  
1000 95% smallest of the set of  $\{z_i\}$ .